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The MSIS and Jacchia-70 (J70) models have been a nearly 55,000 points for which we have measuremedensity between 120 and 200 km. The density meadifferent instruments of five types (ion gauge, spectrometer, calorimeter, and accelerometer) fluoring the 1974-1977 solar minimum. Except for	valuated at each of the nts of the atmospheric surements were made by nine capacitance manometer, mass own on five satellites				

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parameters used in the models (geomagnetic activity, altitude, latitude, local time, season, etc.) were well represented.

The density measurements at any one altitude were found to be consistent with a lognormal distribution; that is, the fractional variations in the density were normally distributed, rather than the variations themselves. If a model correctly predicted the average variation in density, the log of the ratio of the measurements to the predictions should be normally distributed with skewness of 0.0 and kurtosis of 3.0. In fact, the skewness when the J70 model was used was -0.10 and the kurtosis was 3.51. For the MSIS model, the corresponding numbers were -0.06 and 4.42.

The J70 model has systematic errors in both the altitude and local time dependence; when these were corrected, the standard deviation of the log ratio dropped to about 14 percent from about 16 percent. When the MSIS model was used, without any corrections, the standard deviation was 13 percent.

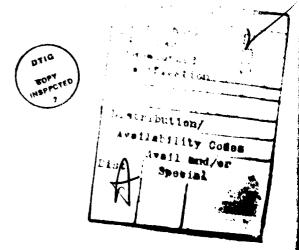
The MSIS model is significantly more accurate than the J70 model, but because the log ratios are not normally distributed, the confidence interval about any one prediction must be estimated empirically rather than analytically. From our data, the probability that the magnitude of the log ratio is less than (0.075, 0.155, 0.205) is (0.5, 0.8, 0.9).

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#### 1.0 SUMMARY AND CONCLUSIONS

#### 1.1 INTRODUCTION

This report documents a comparison that has been made between the Aerospace low altitude atmospheric density data base and two widely-used models, MSIS (Hedin, et al., 1977), and J70 (Jacchia, 1970). By "low altitude" we mean the altitude range between 120 km (the lower limit of the measured densities) and 200 km.

There were five different types of measurements made with nine separate instruments on five satellites. The instruments were ion gauges, capacitance manometers, an accelerometer, a mass spectrometer, and calorimeters. Three of these instruments were flown on the same satellite, AE-E, and so a direct intercomparison allows us to estimate instrumental bias.

#### 1.2 DATA ANALYSIS

The five satellites were all flown in the 1974-1977 period of low solar activity, which limits the generality of the results reported here. However many parameters other than solar flux are of interest, and the data base provides reasonable coverage for all of them. The coverage for latitude, longitude, altitude, geomagnetic index, local time, and month are shown for each instrument (except the

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capacitance manometers, which are included with the ionization gauges) in Section 2.

There are several possible ways of comparing the data to the models; we have chosen to use the logarithm of the ratio of the measurement to the model prediction as the objective function. Figure 1.1 is a histogram of the measured density from AE-C at 175 km, and is typical of all the measurements. It has a pronounced high- density tail, and the two "Pearson parameters" RB1 and B2 are not consistent with a normal distribution. (RB1, (Root Beta 1) is the third moment about the mean divided by the variance raised to the 1.5 power; it is a measure of skewness, and is zero for a normal distribution. B2 (Beta 2) is the fourth moment about the mean divided by the variance squared, and is called the Kurtosis. For a normal distribution it has a value of 3.0. If the kurtosis is larger than this, it indicates that the distribution has an excess number of points near the mean and out in the tails.) The histograms of measured densities at fixed altitudes are consistent with a lognormal distribution, so that the log of the density is normally distributed. (That is, the fractional variations in the density are normally distributed, not the variations themselves.) If the models accurately represent the average variability of the density, we would expect our chosen

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objective function to be normally distributed, and the usual confidence limits would apply.

Section 3 presents histograms of the log ratio for instrument for both models. The MSIS model was slightly modified from that which has been published. The original model used average solar flux parameters and geomagnetic indices Which Were centered at the time of interest; we felt that it was more realistic to use averages that ended at the time of the measurement. Tables 1 and 2 summarize the statistical data. In these tables, AE-C, AE-D, etc. are the satellites, "Density" is the combination of the ion gauges and capacitance manometers, "OSS" is the open-source mass spectrometer, "MESA" is the miniature electrostatic accelerometer, and "Cal" is a calorimeter. The square root of the variances of the standard deviation, RB1, and B2 have been calculated assuming that the underlying distribution is a normal one; in only one case, the AE-E density measurements compared with the J70 model, is RB1 and B2 consistent with a normal distribution.

Figures 1.2 to 1.9 show the mean, standard deviation, skewness and kurtosis for the AE-C density gauges and the AE-E MESA (accelerometer) residuals as a function of altitude. The MSIS model yields a uniformly smaller standard deviation than does J70, but its kurtosis, in

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particular, is much larger, so that an estimate of the confidence interval about any prediction cannot be made other than empirically.

Because the J70 model residuals were more normally distributed, we examined those residuals in more detail, hoping to be able to improve the standard deviation while maintaining the favorable skewness and kurtosis. Section 4 presents plots of the average log ratio versus local solar time for two altitudes, and for both the J70 and the MSIS model. In these figures, the diamonds represent the mean and the error bars through the diamonds are the standard deviation of the mean (not of the individual data points). There are strong, systematic, trends in the U70 residuals, and no obvious trends in the MSIS residuals. Accordingly, we fit the U70 residuals at each altitude with a low-order Fourier time series: a constant term and a diurnal, semi-diurnal, and ter-diurnal cosine. The coefficients were always significant in some altitude range, although no term was significant at all altitudes. The amplitudes and phases so derived are also shown in Section 4, where it can be seen that the semi-diurnal and ter-diurnal terms become increasingly important at lower altitudes and the diurnal term has the opposite behavior. The phase of each of the terms also shows an altitude dependence; perhaps the most

surprising is the shift of the diurnal bulge to early morning at the lower altitudes.

applied to the very same data that were used to derive them, we would expect to see a significant change in the standard deviation of the corrected residuals. Histograms of the "before" and "after" distributions are shown in the final part of Section 4. While there is a (statistically) significant reduction in the standard deviation, it is still larger than that obtained using the MSIS model with no corrections at all.

#### 1.3 CONCLUSIONS

We have compared over 50,000 separate measurements of atmospheric density between 120 km and 200 km with the predictions of two models, the MSIS and the J70. The MSIS model produces a considerably smaller standard deviation of the residuals than does the J70, even after the observed systematic errors in local time and altitude have been removed. The kurtosis of the MSIS residuals is large: this may indicate that there are still nonrandom effects that this model has failed to capture, so there is still hope that it may be significantly improved in the future. Until that time, however, we have no choice but to use empirical cumulative probability functions to estimate confidence

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limits of the accuracy of its predictions. These are shown in Figures 1.10 to 1.17.

The average bias of the log ratio is near zero for both models (cf. Tables 1 and 2), and, in particular, is near zero for the three instruments on AE-E considered separately. This indicates that a major portion of the average error is due to instrumental bias rather than model bias, so there is justification in subtracting the individual biases and amalgamating the data. The cumulative empirical probability function for all the data using the MSIS model is shown in Figure 1.18, and represents our best estimate of the true distribution function for times near solar minimum.

#### 1.4 REFERENCES

Hedin, A. E., et al., "A Global Thermospheric Model Based on Mass Spectrometer and Incoherent Scatter Data, 1, N2 Density and Temperature", J. Geophys. Res. 82, 2139-2147, 1977.

Hedin, A. E., et al., "A Global Thermospheric Model Based on Mass spectrometer and Incoherent Scatter Data MSIS, 2, Composition", J. Geophys. Res. 82, 2148-2156, 1977.

Jacchia, L. G., "New Static Models of the Thermosphere and Exosphere with Empirical Temperature Profiles", Smithsonian Astrophys. Obs. Special Report No. 313, 87 pp., 1970.

### 2.0 GEOPHYSICAL PARAMETER COVERAGE

### 2.1 ALTITUDE DISTRIBUTION

Figures 2.1 through 2.8 show the altitude coverage of the eight instruments (the ion gauges and the capacitance manometers are considered as one instrument). Only the AE-C density gauges and the calorimeters have a significant number of points below 135 km, and only Calorimeter-3 measured data below 125 km. The combined altitude distribution is given in Fig. 2.9. The total number of points, 54,506, is different from the numbers given in Tables One and Two of Section 1 because these data were not aditted; measurements were retained in Section One only if the log of the ratio (measurement/model) were between plus and minus one.

#### 2.2 LATITUDE DISTRIBUTION

Figures 2.10 through 2.17 show the latitude distribution of the measurements. The northern hemisphere is well covered, as is the equatorial region (AE-E had an inclination of 19.6 degrees), but only AE-C and AE-D collected a significant amount of data below 20 S.

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#### 2.3 LONGITUDE DISTRIBUTION

Figures 2.18 through 2.25 show the longitude distribution of the measurements. Overall, there is excellent longitude coverage.

### 2.4 LOCAL TIME DISTRIBUTION

Figures 2.26 through 2.33 show the local time distribution of the measurements. AE-C and AE-E have nearly uniform coverage; the measurements on AE-D are concentrated in the morning and evening sectors. S3-1 and Calorimeter-3 made measurements primarily in the late morning and afternoon, while the data from Calorimeter-4 is almost exclusively between 1100 and 1300.

## 2.5 MONTHLY DISTRIBUTION

Figures 2.34 through 2.41 show the monthly distribution of the measurements. AE-C and AE-E cover all four seasons, AE-D took data between September and January, \$3-1 was useful primarily in October, Calorimeter-3 in March and April, and Calorimeter-4 in September and October.

### 2.6 24-HOUR AP DISTRIBUTION

Figures 2.42 through 2.48 show the distribution of the 24-hour average Ap index. These averages ended at the time of each measurement, and are used in the MSIS model

only. This index has a maximum possible value of 400; the Figures show that its values in this data base are stongly concentrated in the lower twenty percent of its range. This is consistent with the fact that this is a sunspot-minimum data base. (Even during sunspot-maximum periods, geomagnetic indices are more frequently in the lower portion of their range. See the discussion in the following sub-section.)

#### 2.7 KP DISTRIBUTION

Figures 2.50 through 2.57 show the distribution of the three-hour geomegnetic index Kp. This index was evaluated six hours prior to each density measurement, and is used only in the J70 model. As with the 24-hour average Ap, most of the data were taken when Kp(6) was in the lower portion of its 0-9 range. The effect is not as dramatic here, however, because Ap and Kp are logarithmically related. Figure 2.58 shows the Kp distribution from 1 January 1932 to 30 June 1981, and so includes periods of both high and low solar activity. Its distribution is not unlike the Kp distribution when the density measurements were made. The average value over this 80 year period was 2.12, with a standard deviation of 1.46. This corresponds to an average Ap of 7.7 (plus 13.3, minus 4.7). Although there were no great geomagnetic storms during the time for

which we have data, this is not considered to be operationally significant. (Kp reached its maximum value of 3 only 20 times in 50 years, and exceeded 6+ (Ap equals 34) only one percent of the time.)

#### 3.0 MEASUREMENT/MODEL DISTRIBUTIONS

#### 3.1 THE MSIS MODEL

Figures 3.1 through 3.8 are histograms of the natural log of the ratio of the individual measurements to the predictions of the MSIS model. The data were editted so that the magnitude of this ratio was always between zero and one; this eliminated 28 points (out of 54,606). Most of the editted points were obviously in error: a reported measurement of 1E-3 grams/centimeter-cubed, for instance, instead of 1E-13.

The MSIS model was avaluated using 24-hour average values of Ap and 81-day average values of the solar flux parameter F10.7 ending at the time of the measurement. The original model specified that these averages were to be centered at the time of interest, but it was difficult for us to see the physical justification for that procedure. In fact, histograms comparable to the ones presented in this subsection were generated using the original model; in every case, the standard deviations were slightly larger.

#### 3.2 THE J70 MODEL

Figures 3.8 through 3.16 are histograms of the natural log of the individual density measurements divided by the predictions of the J70 model. As in the previous

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subsection, these data were editted to exclude points with magnitude greater than one. A single additional measurement was eliminated when the U70 model was used.

#### 4.0 CORRECTIONS TO THE J70 MODEL

#### 4.1 LOCAL TIME DEPENDENCE

Figures 4.1 and 4.2 show the local time dependence of the natural log of the AE-C density gauge measurements between 195 and 200 km divided by the predictions of the J70 and the MSIS models, respectively. Figures 4.3 and 4.4 show the same thing for the AE-E MESA measurements. Figures 4.5 to 4.8 are comparable to the first four, but for the altitude range 150-155 km. It can be seen that the J70 plots exhibit a persistent, reasonably strong, local time dependence for both instruments in both altitude ranges. The MSIS plot at the higher altitude for AE-C also seems to show a local time dependence, but it does not persist across instruments nor at the lower altitude.

We made an empirical model of the error in the time dependence of the J70 model using the AE-C density gauge data. In each five kilometer interval, the log ratio was fit to a third-order Fourier expansion, using a constant term and diurnal, semi-diurnal, and ter-diurnal terms. The coefficients were determined using a weighted least- squares technique, the weights being the reciprocal of the variances of the means at each local time. The altitude dependence of the coefficients is shown in Figures 4.9 through 4.15. The phases are determined such that a negative phase indicates

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that the model's prediction should be increased at later times. For instance, the -44 degree phase angle for the diurnal term at 165 km, coupled with the 0.092 value for the amplitude, indicates that the model's prediction should be increased by nine percent at about 0300 and decreased by the same amount at 1500.

These corrections were applied to the J70 model, and the histogram of the log ratios was recalculated. The original histogram is shown in Figure 4.16 and the "corrected" one in Figure 4.17. The standard deviation has been reduced from 0.156 to 0.142, a statistically significant amount. It is not operationally significant, however, since the standard deviation using the uncorrected MSIS model is substantially smaller, 0.126, as is shown in Figure 4.18.

Since the correction procedure did not produce very dramatic improvements for the AE-C data, we did not feel it to be worthwhile to pursue the matter with the other data sets.

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TABLE 1
MSIS STATISTICS

	N	Mu	Sigma	Vvar(sig)	RB1	(Var(RB1)	B2 √v	ar(82)
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AE-C Density	17415	0.055	0.126	0.001	-0.304	0.030	5.246	0.135
AE-D Density	10935	-0.090	0.132	0.001	0.120	0.037	3.243	0.128
AE-E Density	4977	0.089	0.130	0.001	-0.098	0.055	3.123	0.187
AE-E DSS	7574	-0.058	0.132	0.001	-0.054	0.045	4.589	0.187
AE-E MESA	8280	-0.006	0.127	0.001	0.035	0.043	5.140	0.193
53-1 Density	1454	0.100	0.144	0.003	0.289	0.109	3.088	0.344
Cal-3	1492	-0.018	0.174	0.003	0.256	0.101	4.849	0.436
Ce1-4	2451	0.145	0.136	0.002	0.201	0.079	3.919	0.299
All Date	54578	0.008	0.130	0.007	-0.058	0.189	4.180	0.920

TABLE 2 J70 STATISTICS

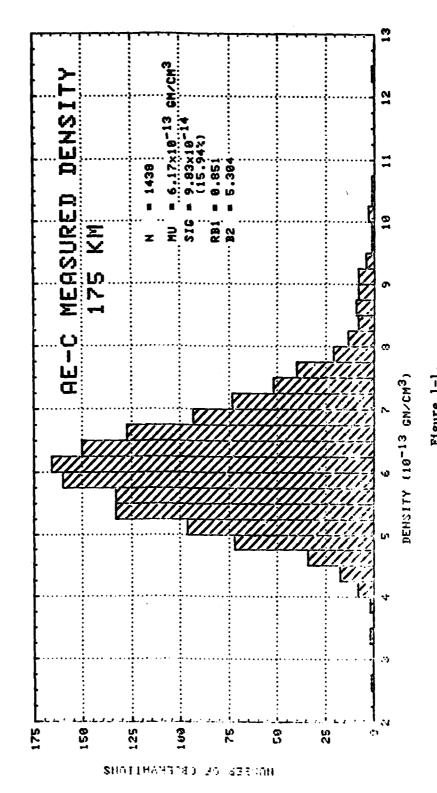
	N	Mu	Sigma	Vvar(sig)	RB1	/var(RB1)	82 /	/ar(B2)
	-						•	•
AE-C Density	17415	0.050	0.156	0.001	-0.298	0.030	3.935	0.112
AE-D Density	10935	-0.113	0.147	0.001	-0.064	0.037	3.006	0.124
AE-E density	4977	0.084	0.160	0.002	-0.008	0.055	2.924	0.182
AE-E OSS	7573	-0.063	0.170	0.001	-0.128	0.045	3.253	0.154
AE-E MESA	8280	-0.017	0.156	0.001	-0.011	0.043	3.763	0.159
\$3-1 Density	1454	0.012	0.138	0.003	0.412	0.104	3.528	0.367
Cal-3	1492	-0.004	0.192	0.004	0.324	0.102	4.171	0.397
Ca1-4	2451	0.010	0.129	0.002	0.212	0.079	3.545	0.283
All Data	<b>5</b> 4577	-0.011	0.156	0.011	-0.097	0.175	3.462	0.415

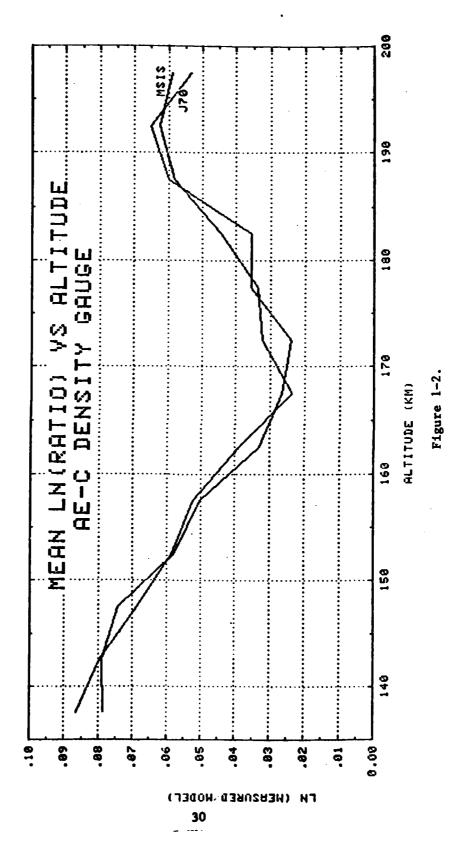
Tables 1 and 2

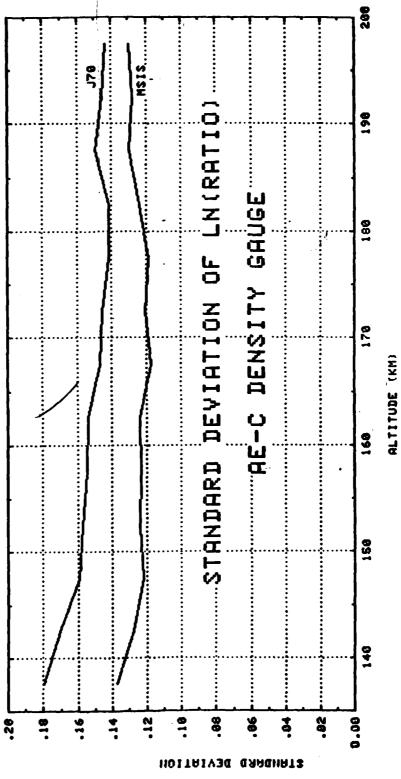
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SECTION 1

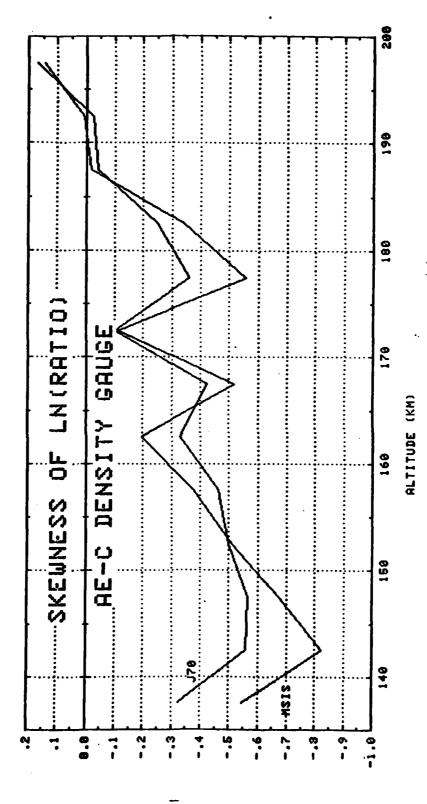
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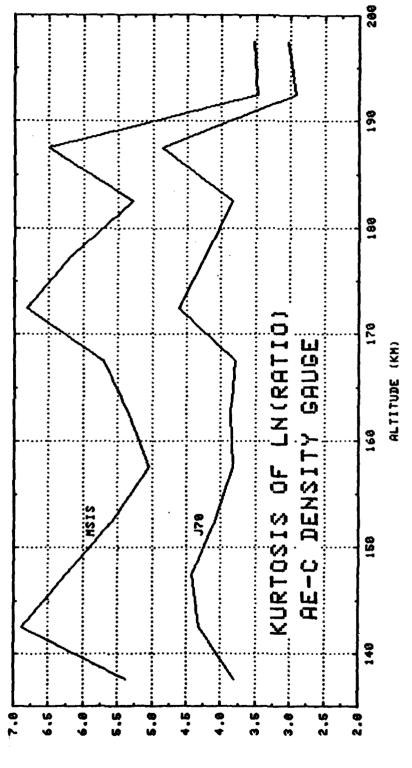


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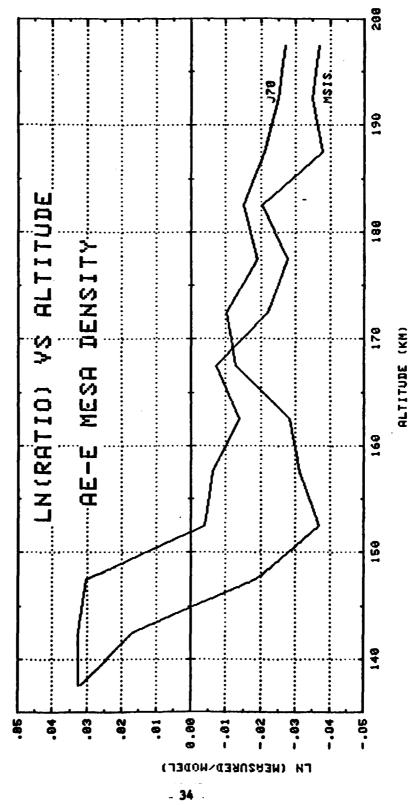


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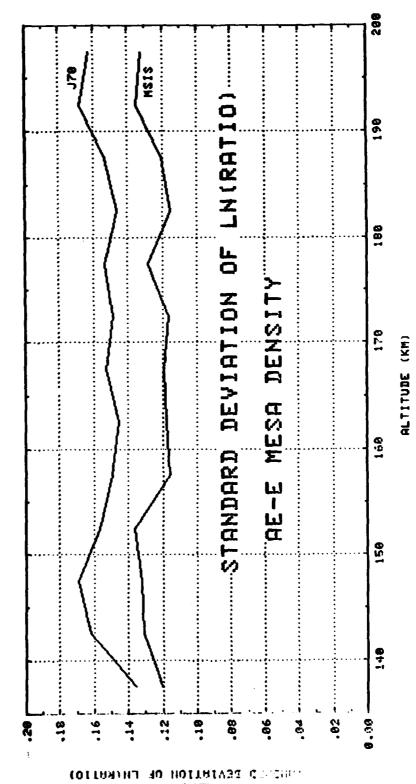
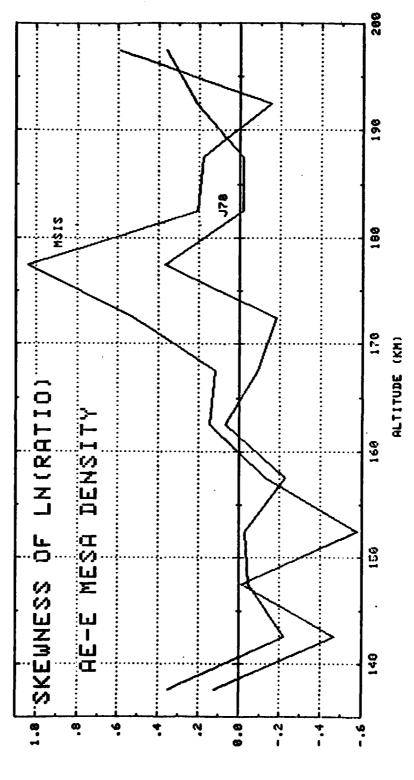
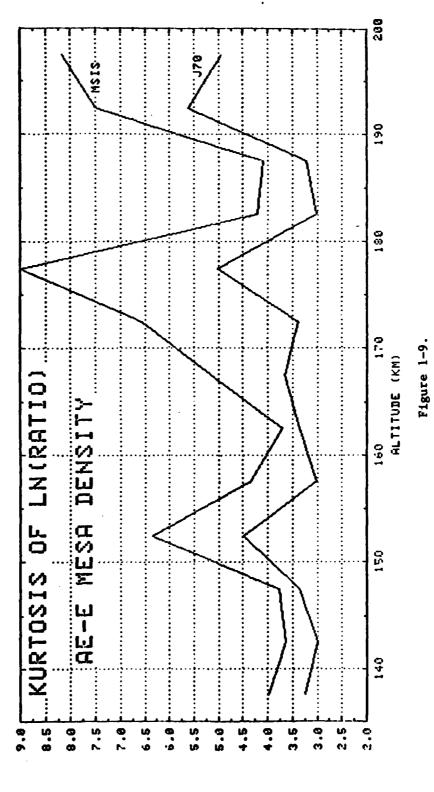
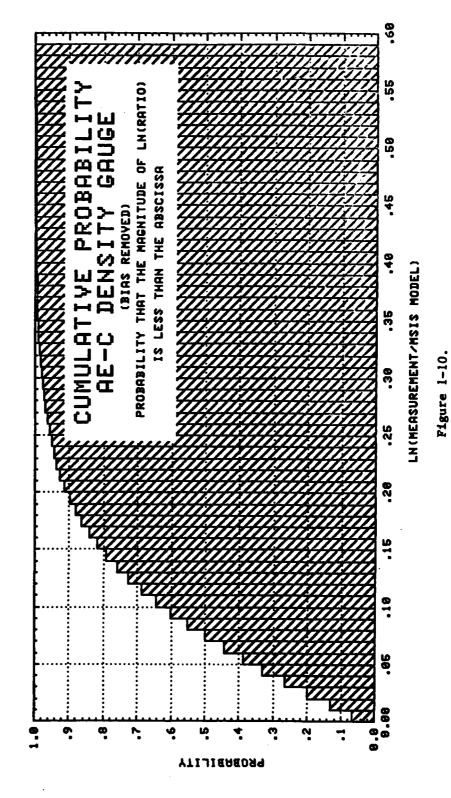


Figure 1-7.







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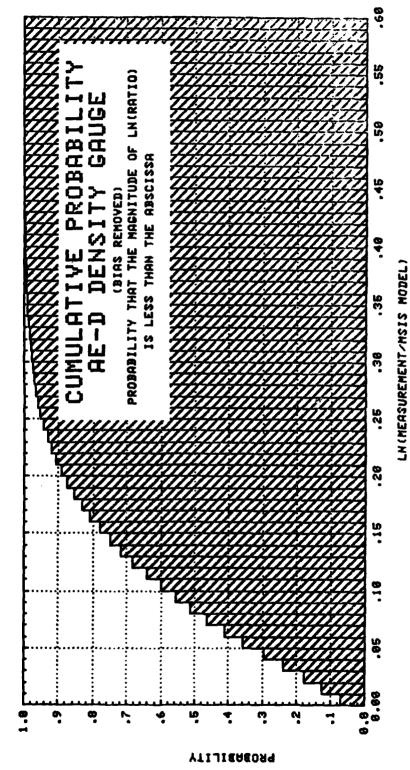
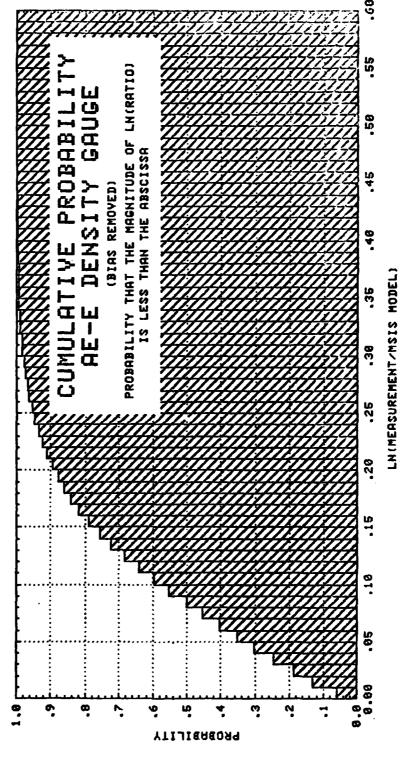
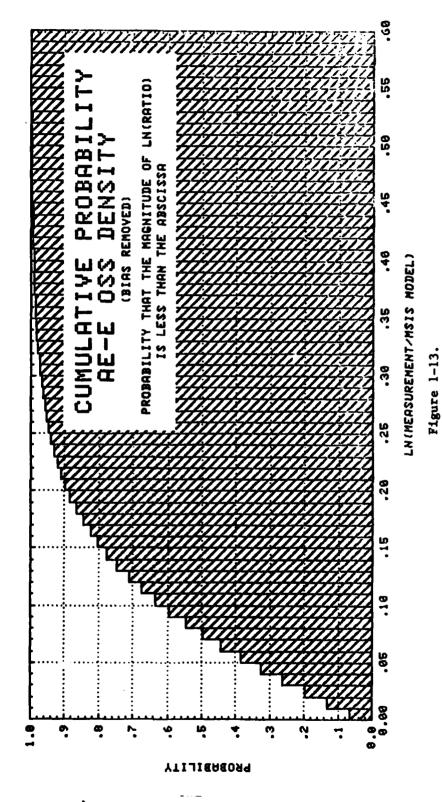
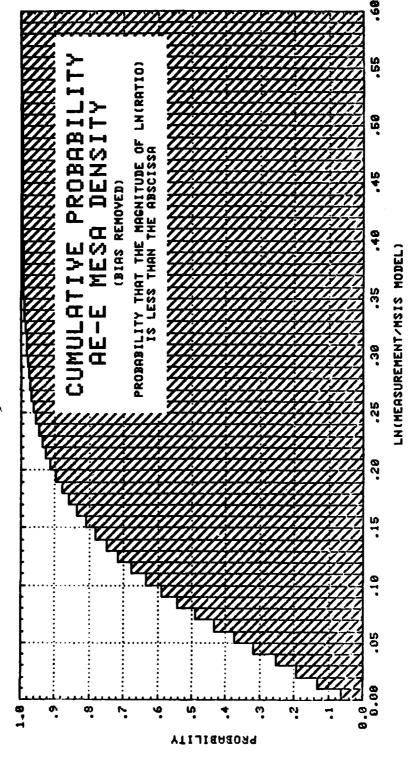
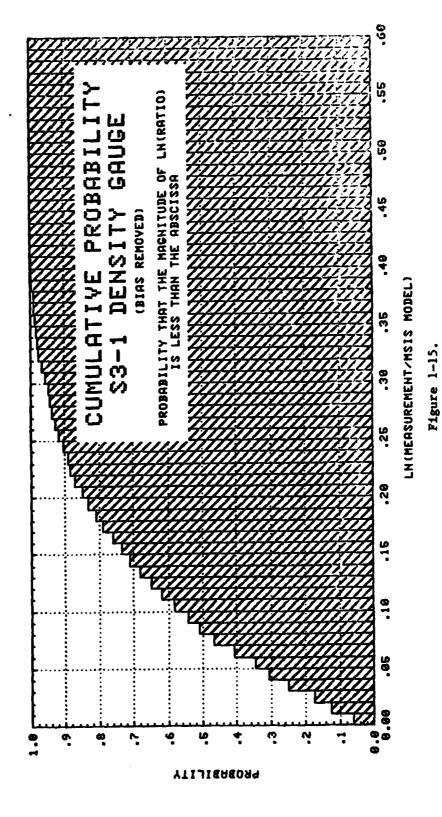


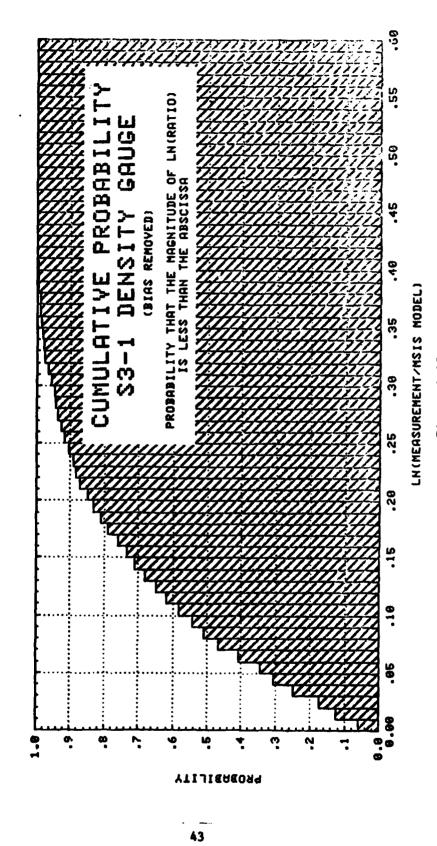
Figure 1-11.











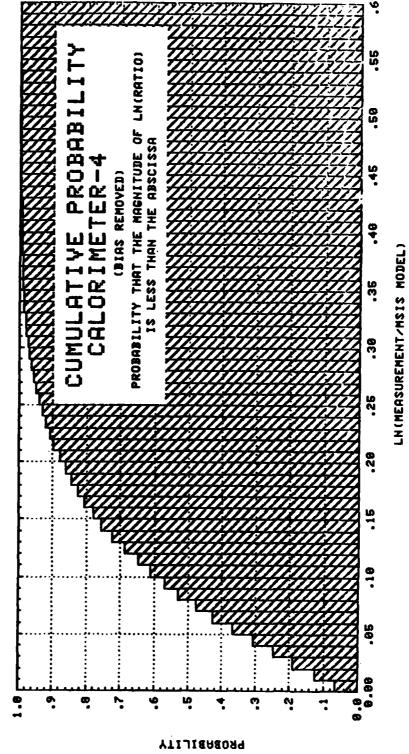


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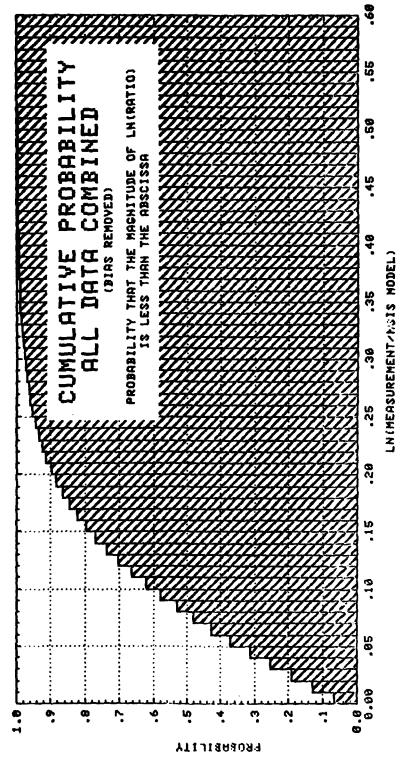
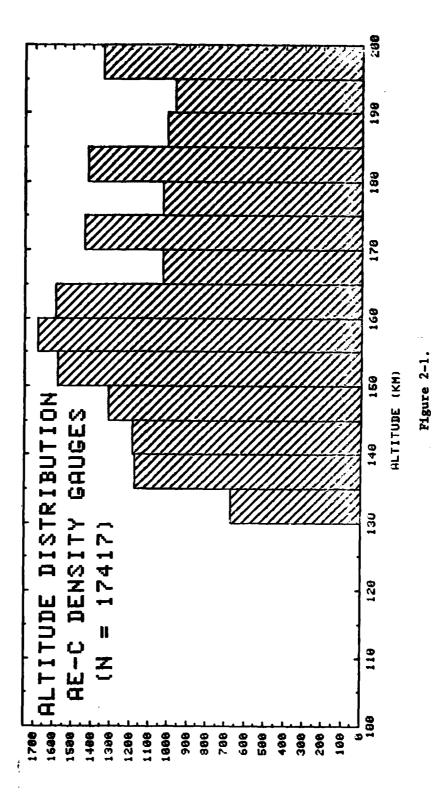


Figure 1-18.

## SECTION 2



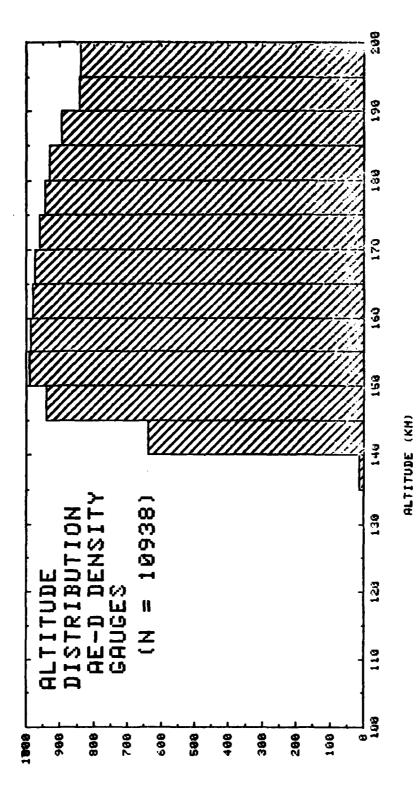
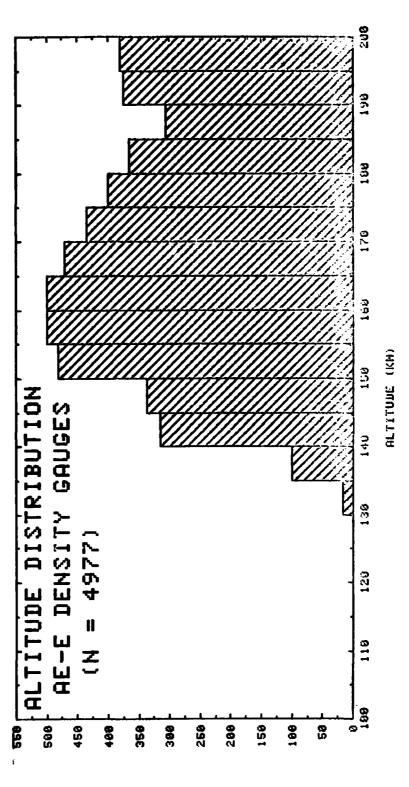


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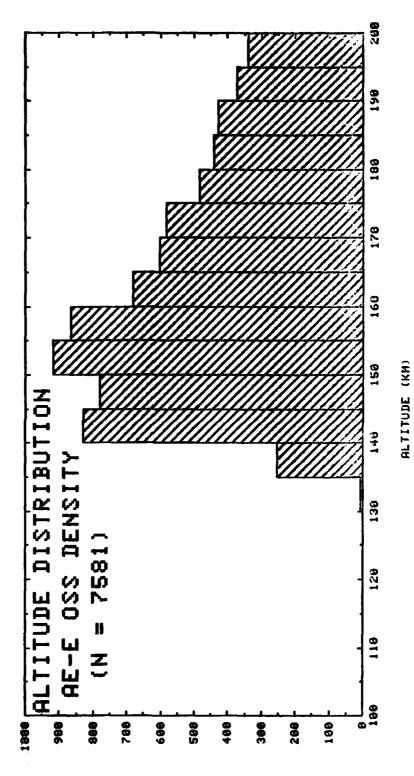


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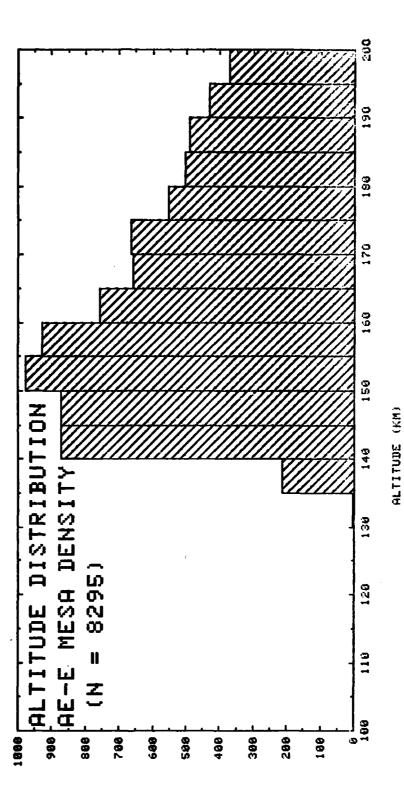


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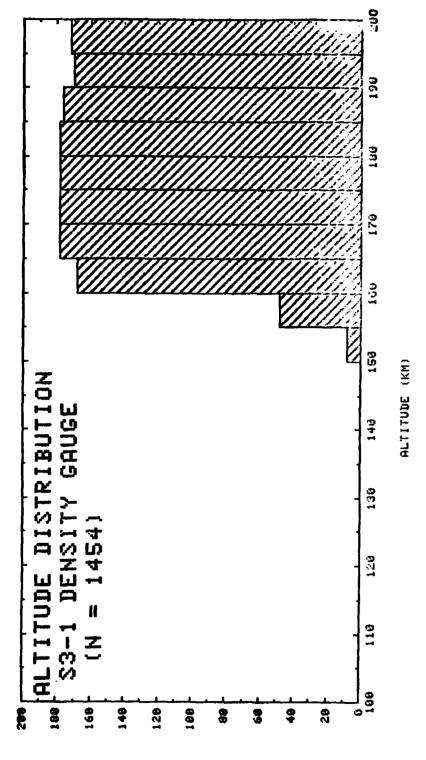


Figure 2-6.

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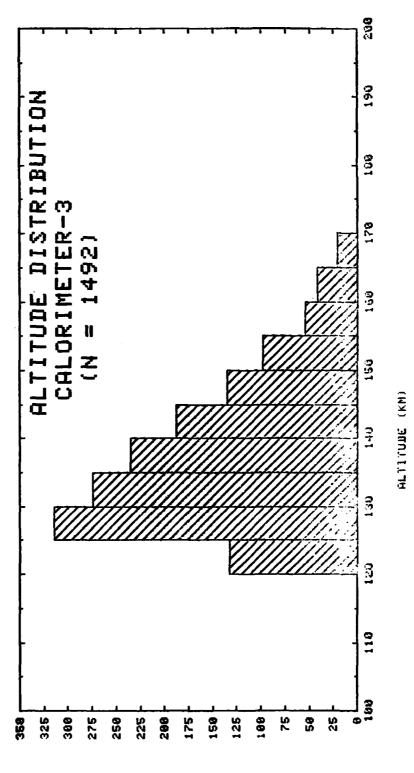


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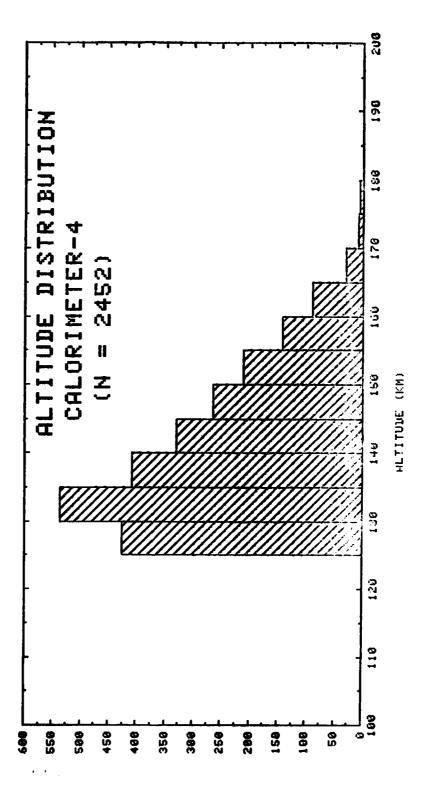
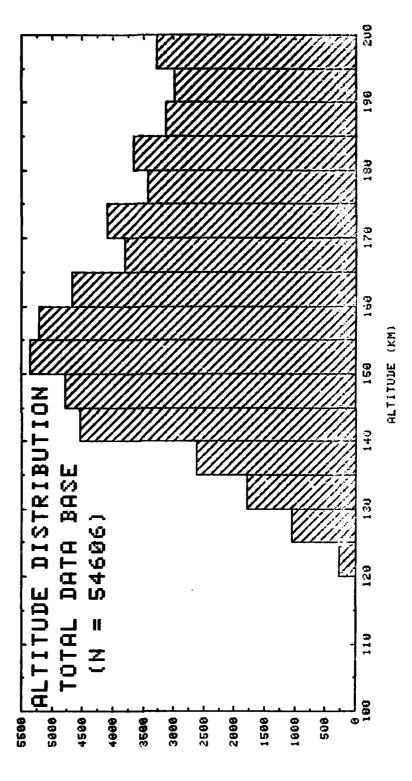
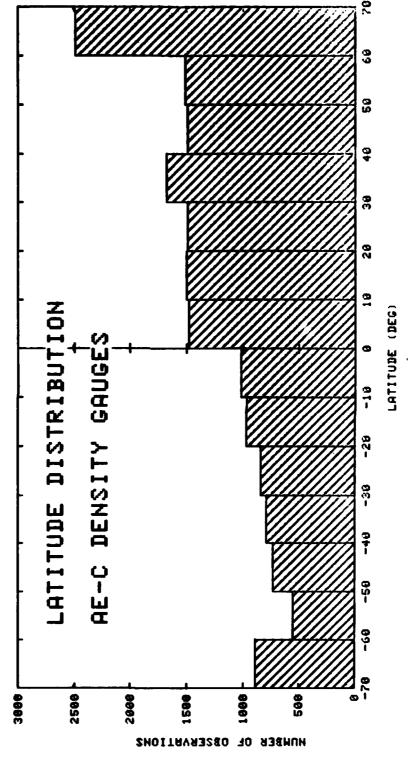
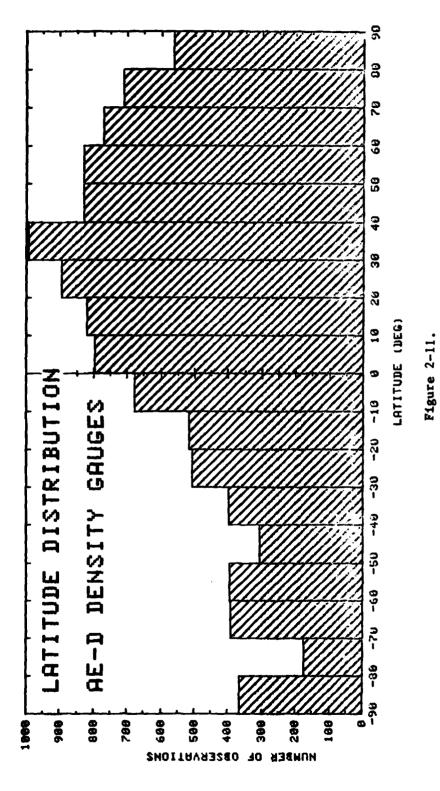
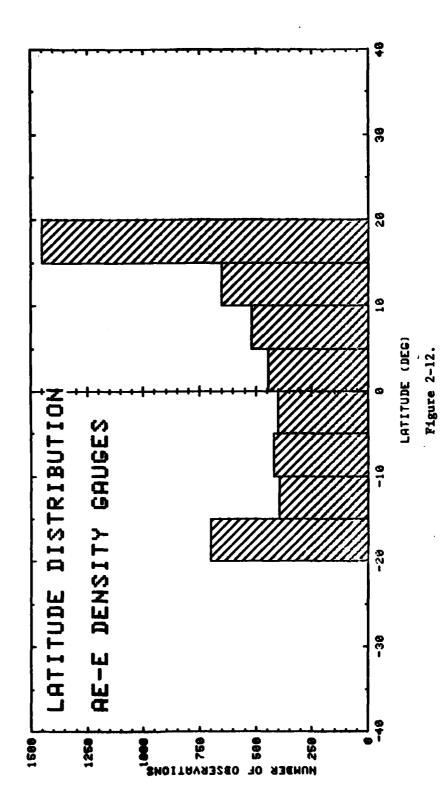


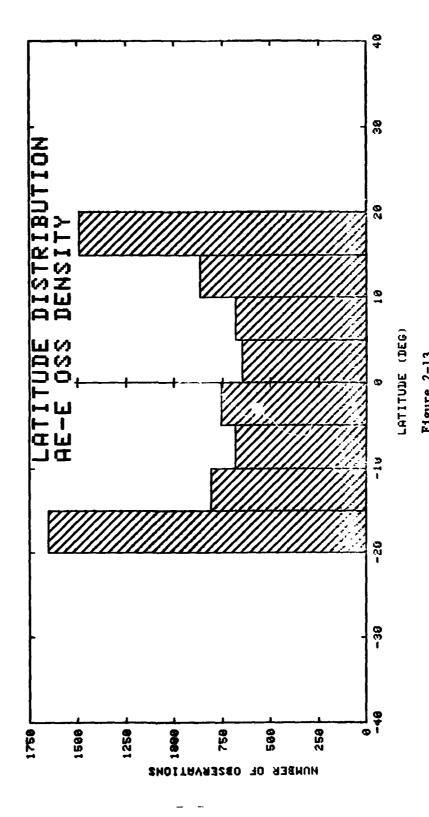
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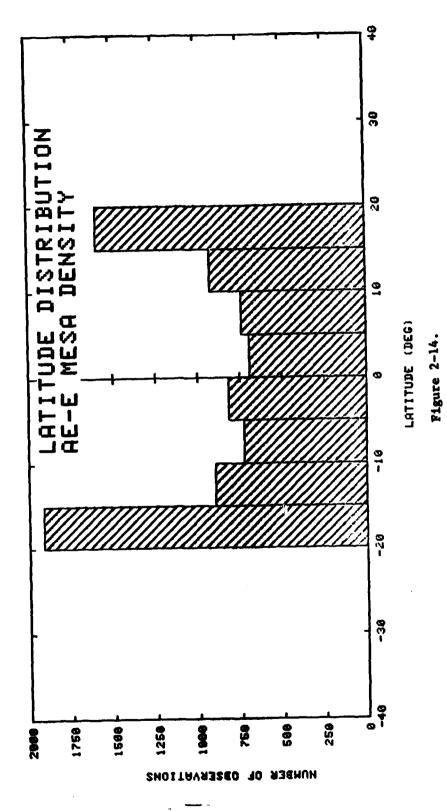


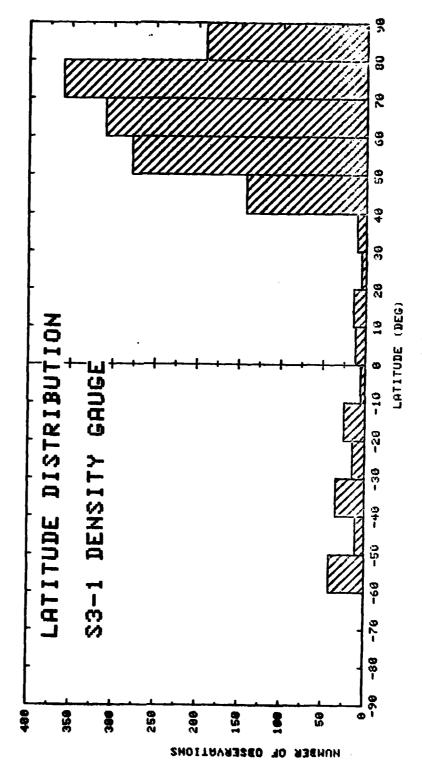


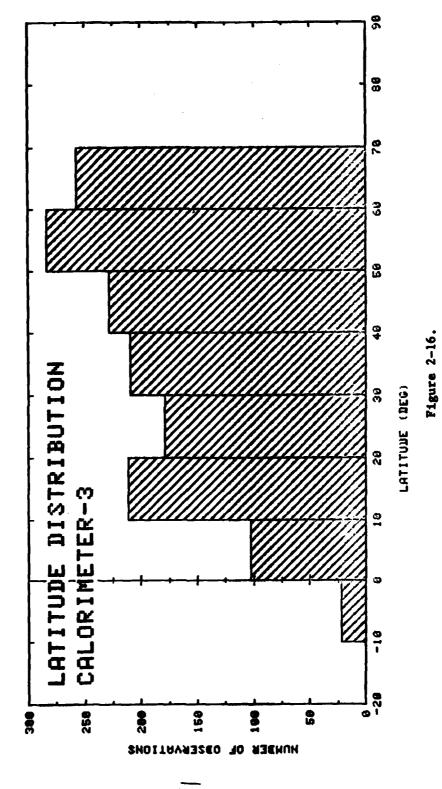


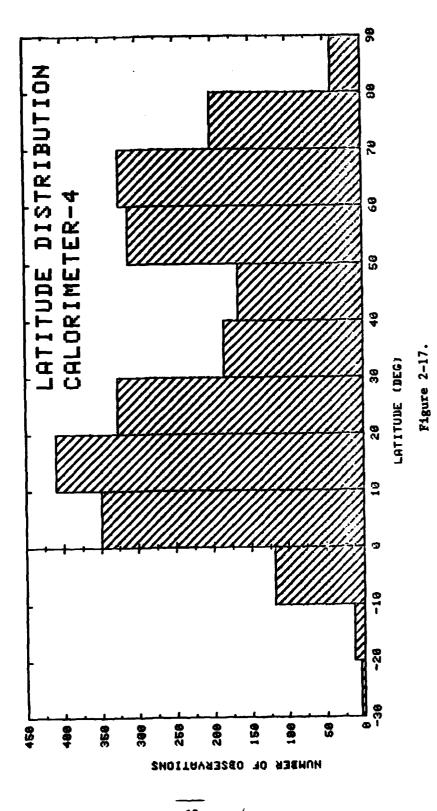


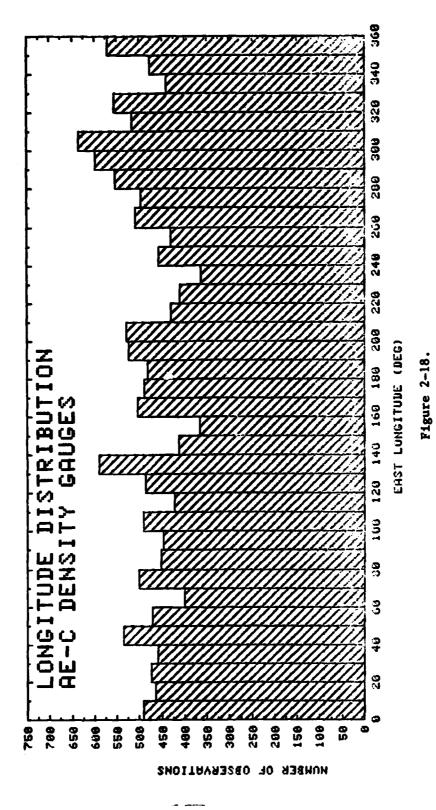












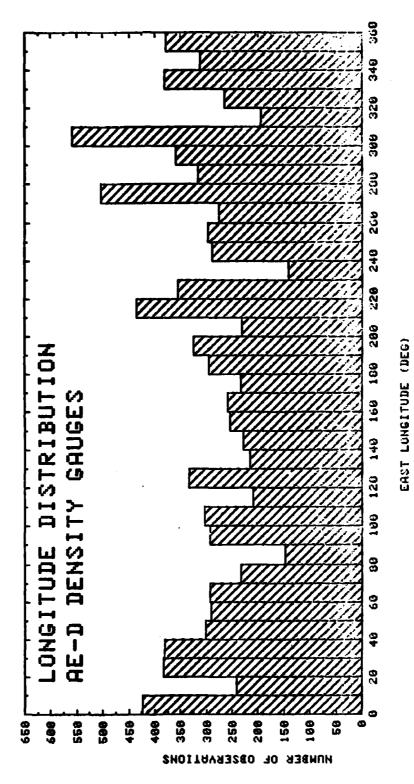
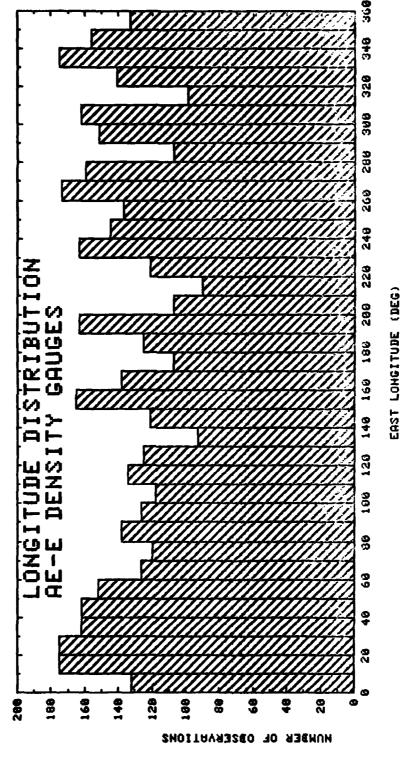
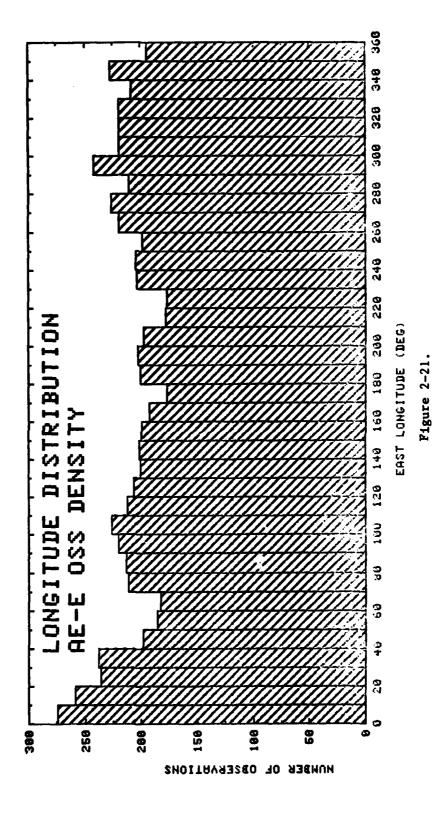
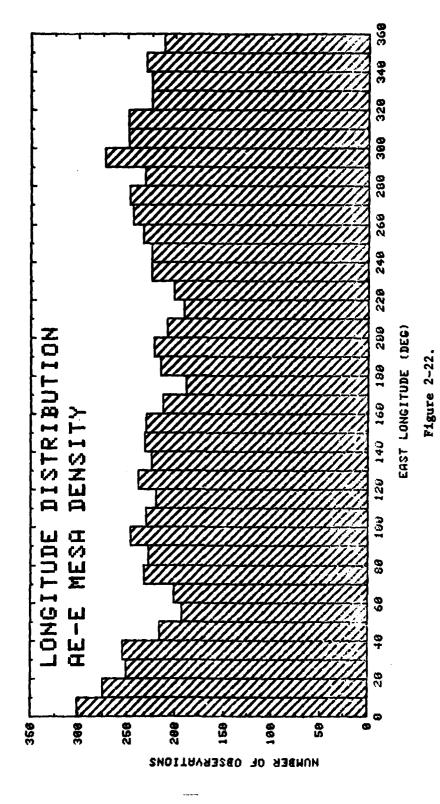
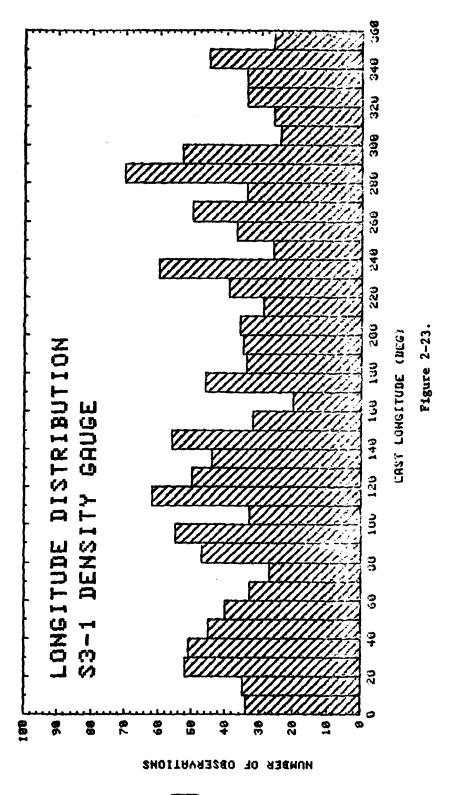


Figure 2-19.









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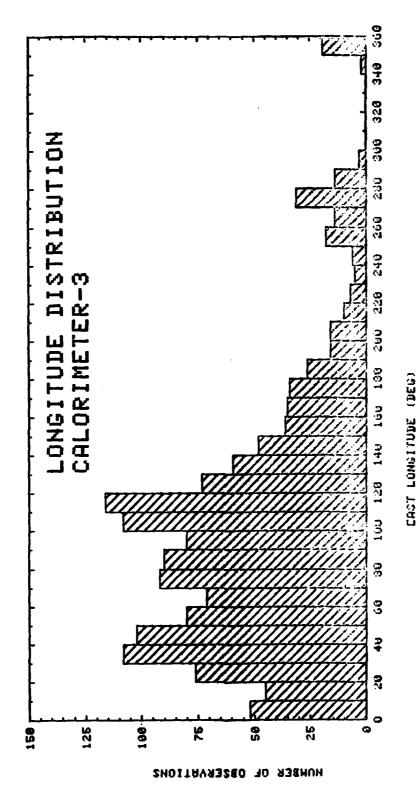


Figure 2-24.

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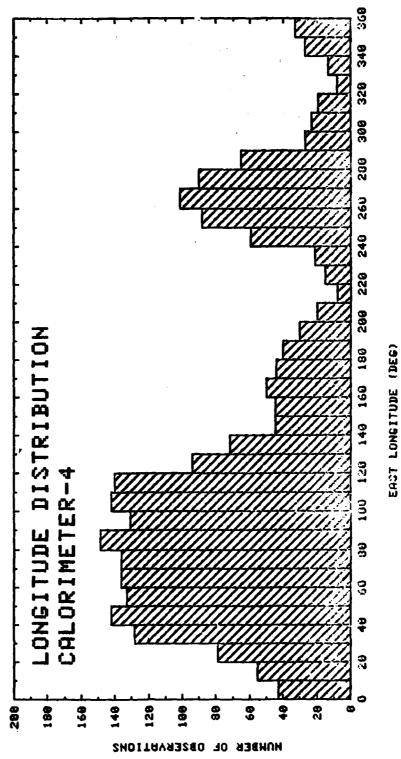
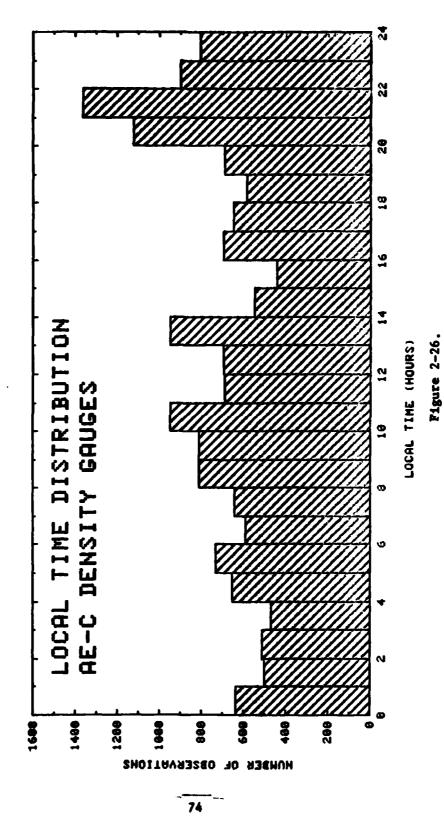


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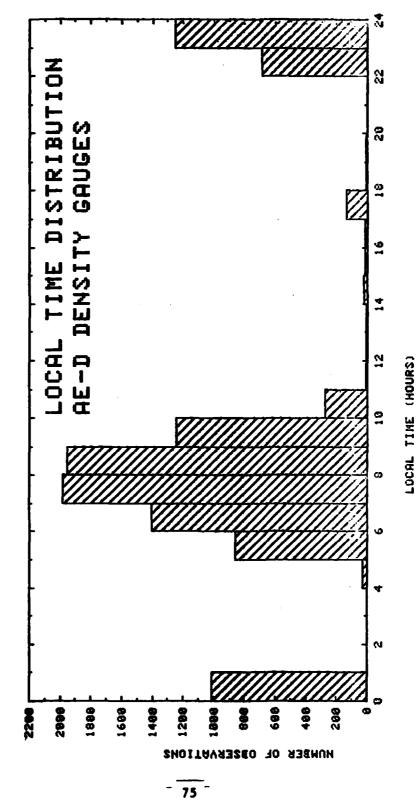


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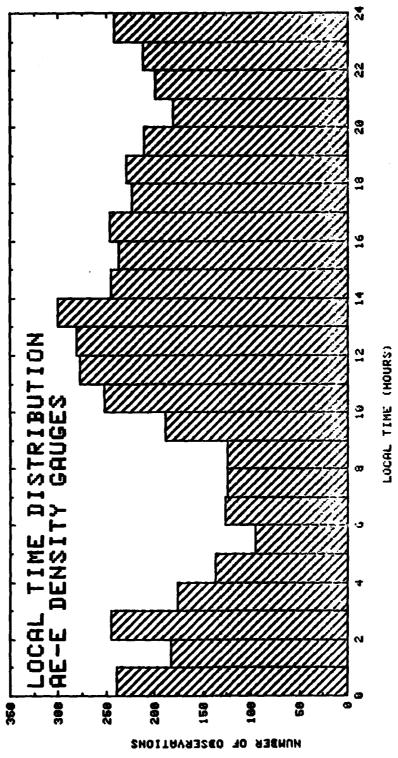
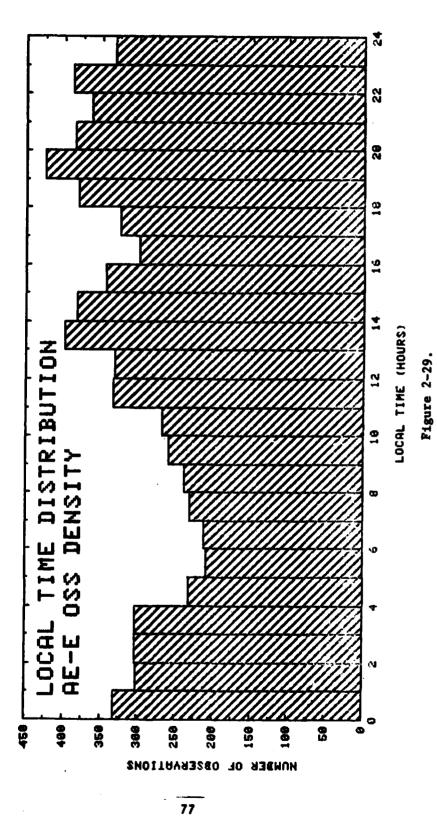
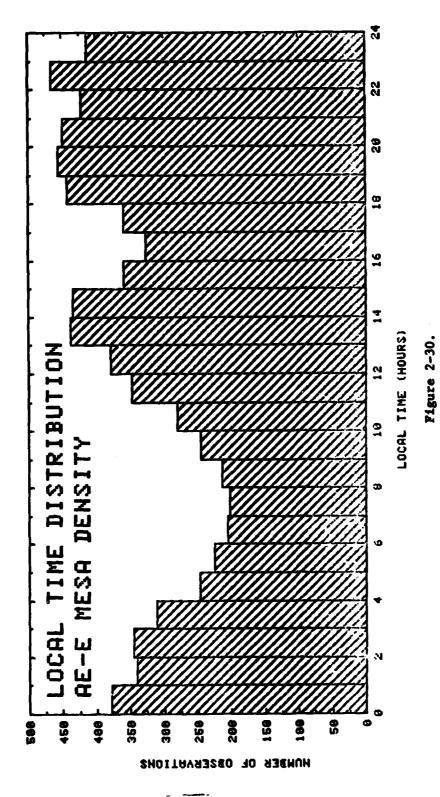


Figure 2-28.

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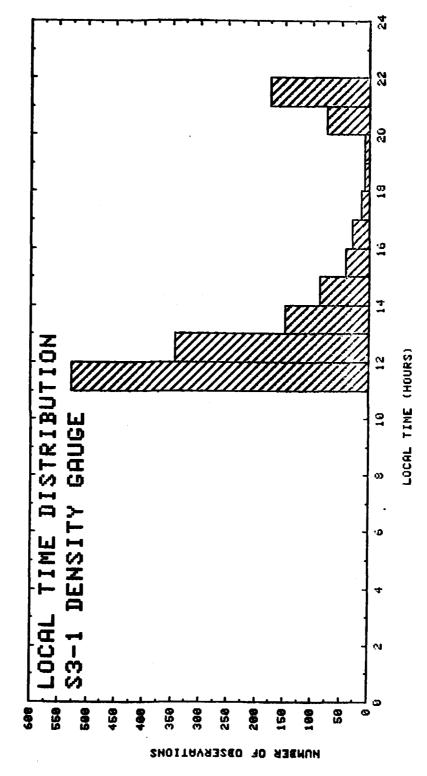
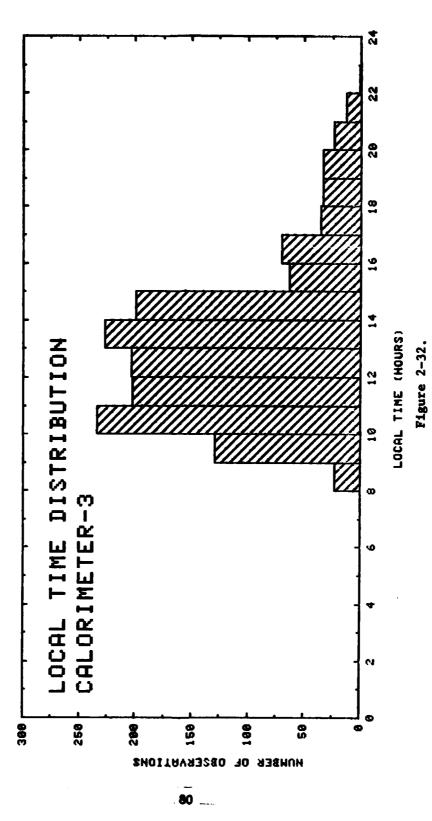
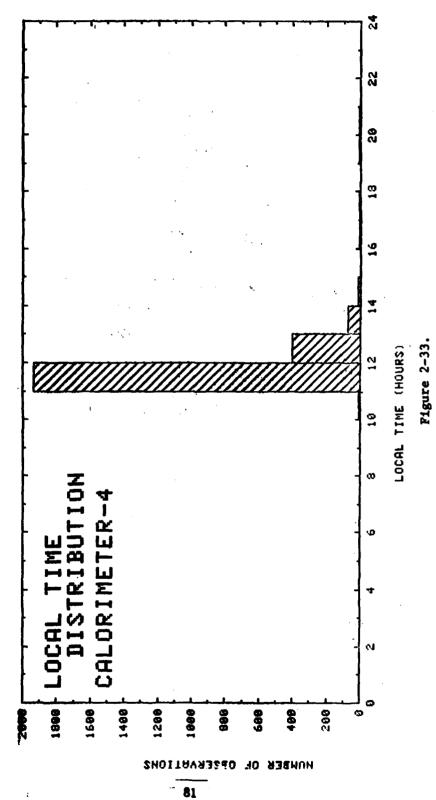
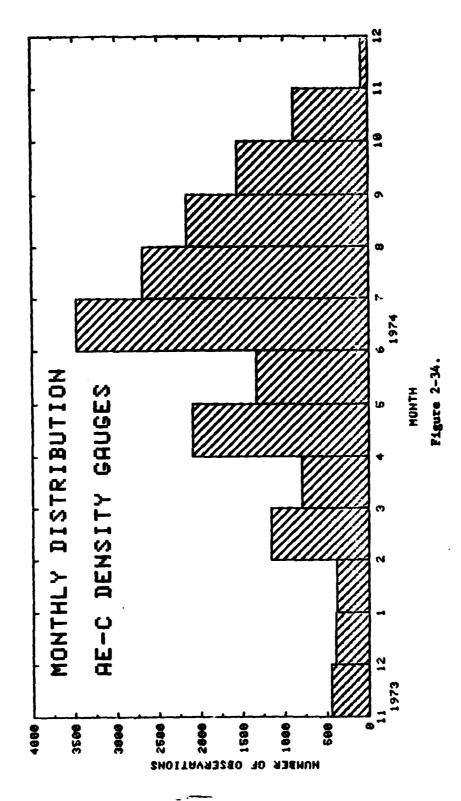


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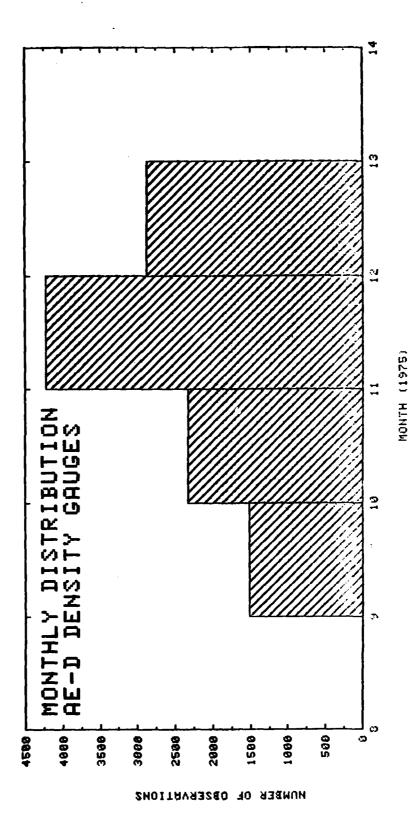
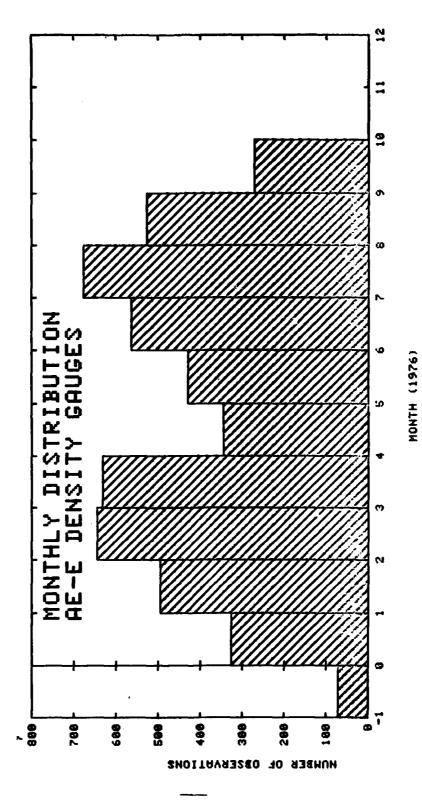
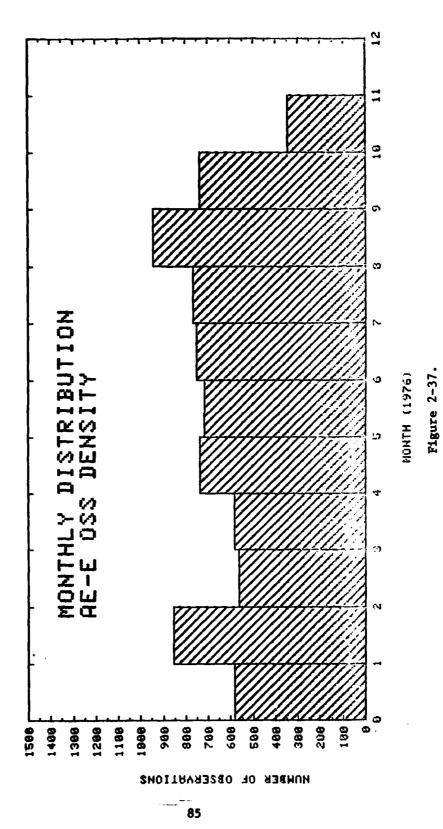


Figure 2-35.

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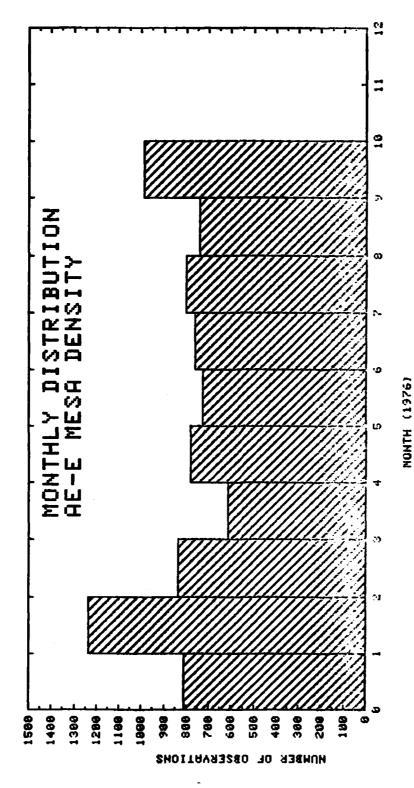


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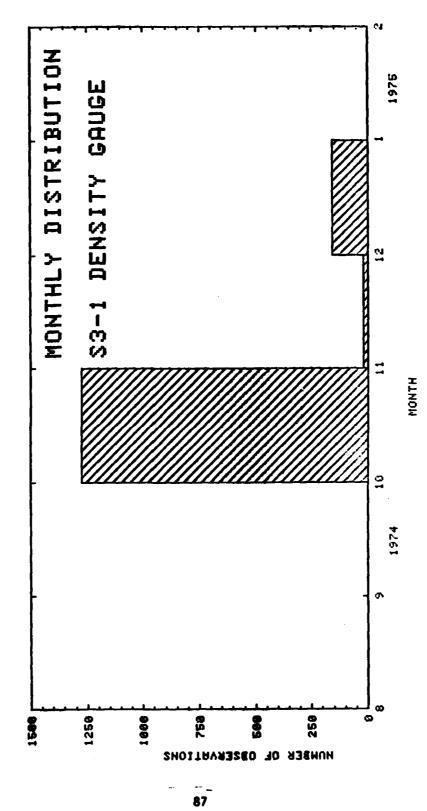


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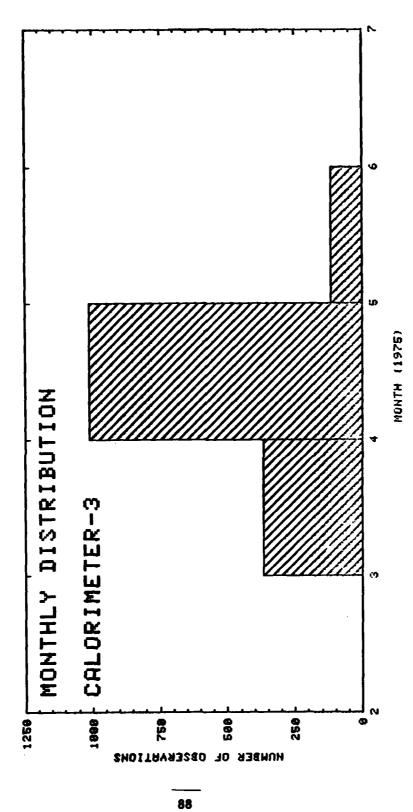


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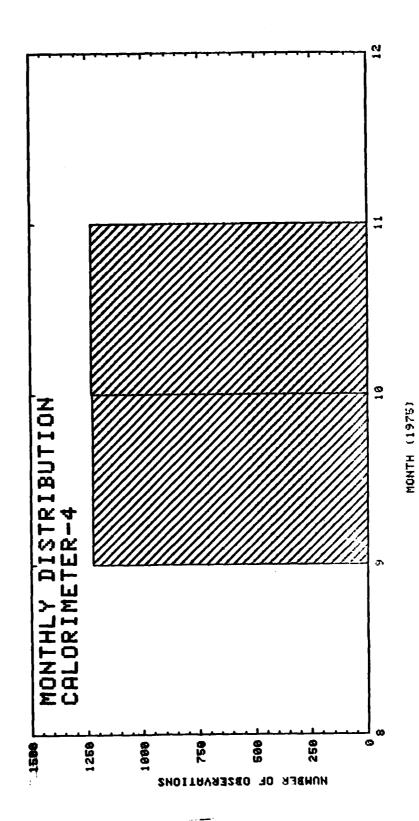
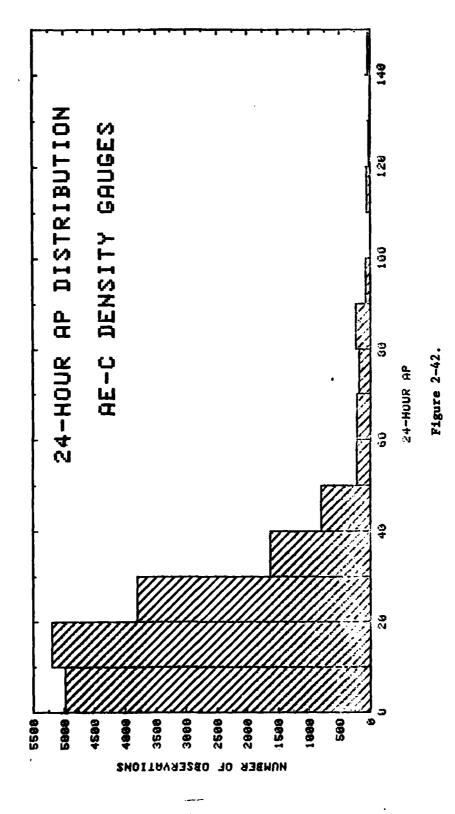
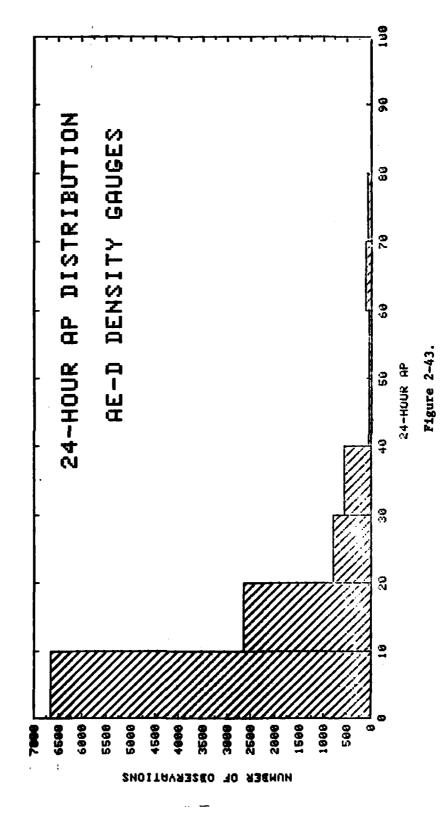
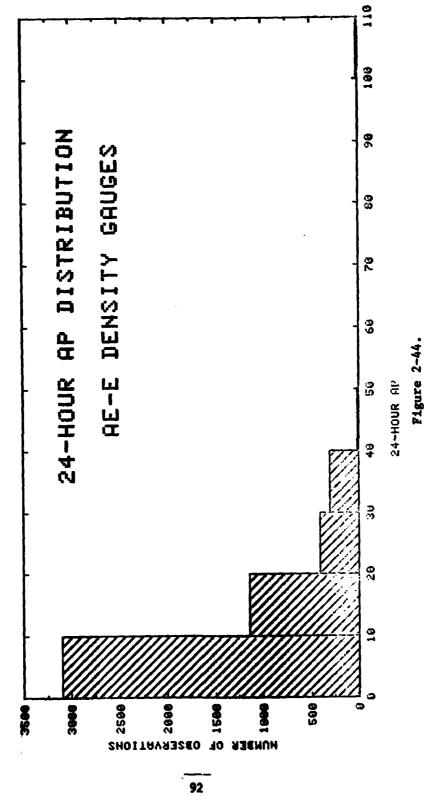
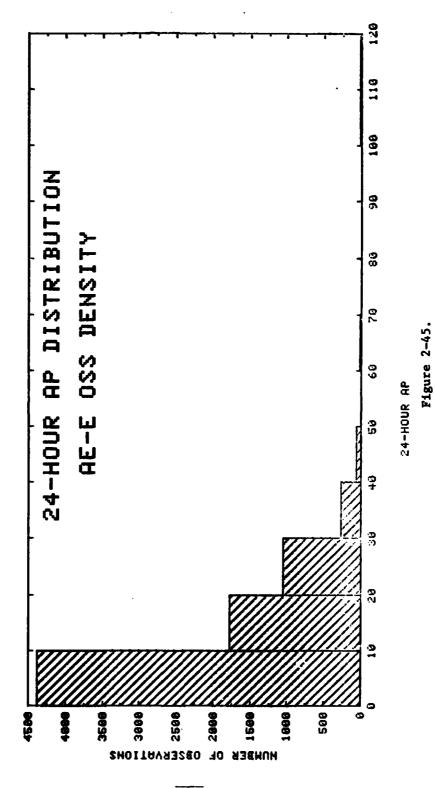


Figure 2-41.









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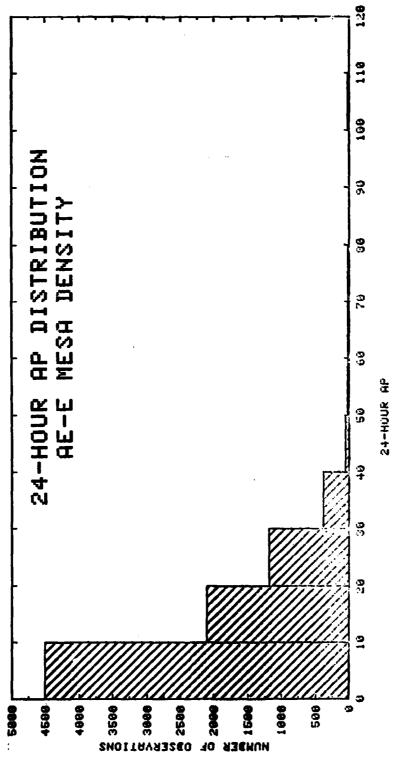
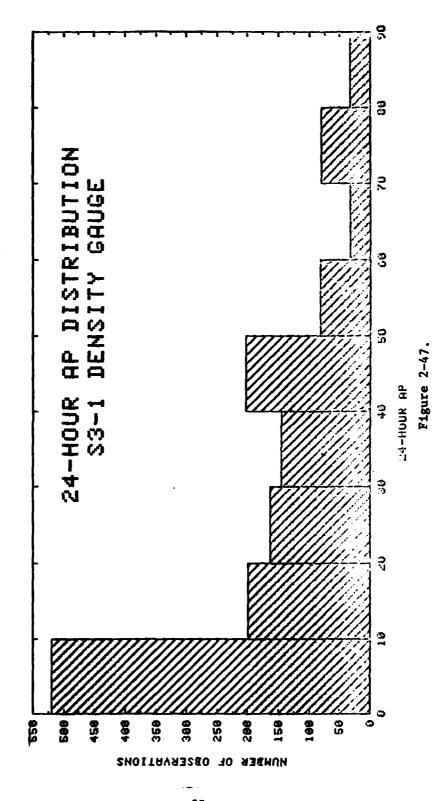


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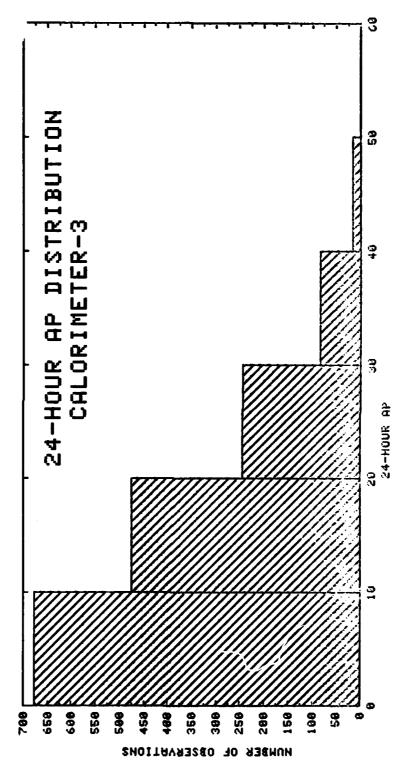
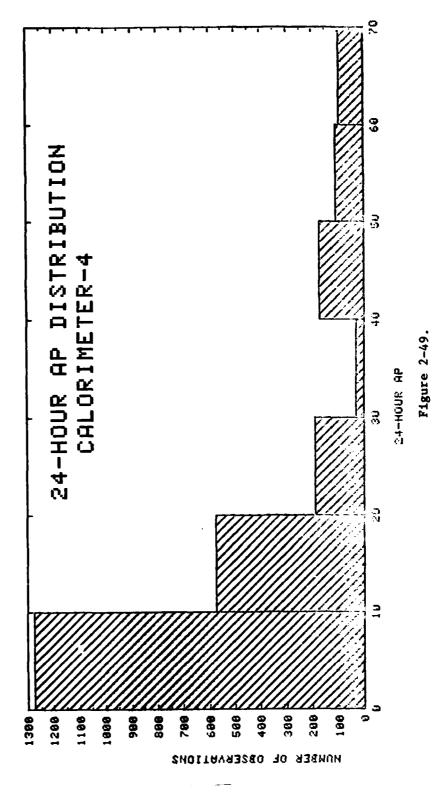
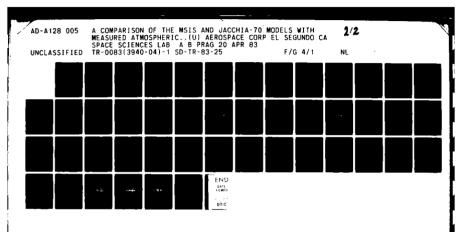


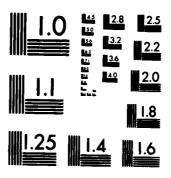
Figure 2-48.

Advantage of Barrier



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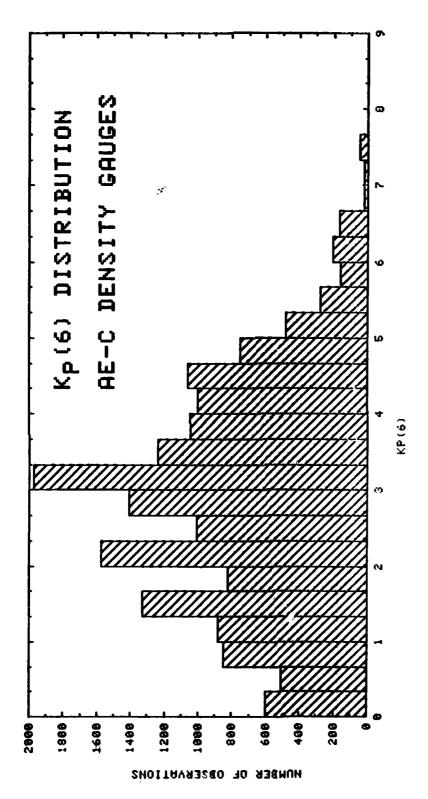
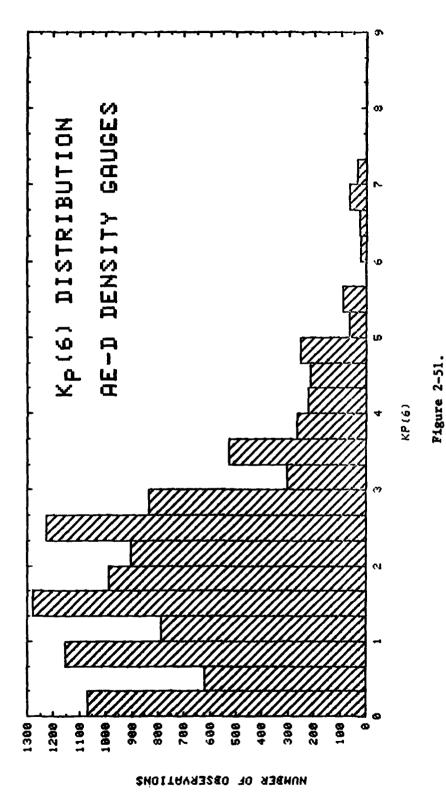
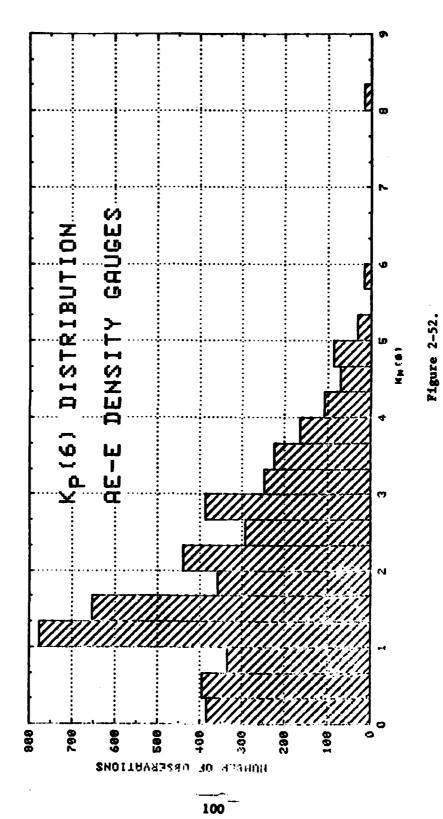
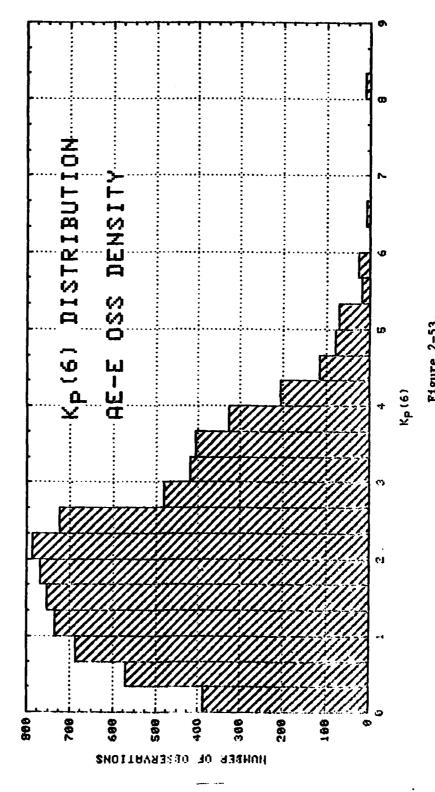
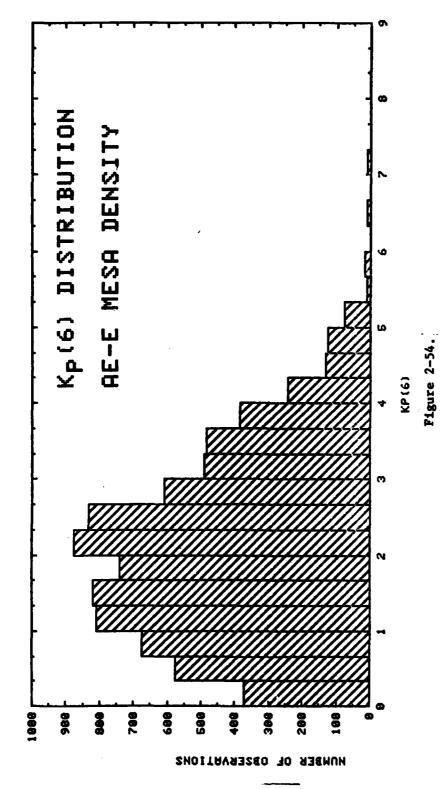


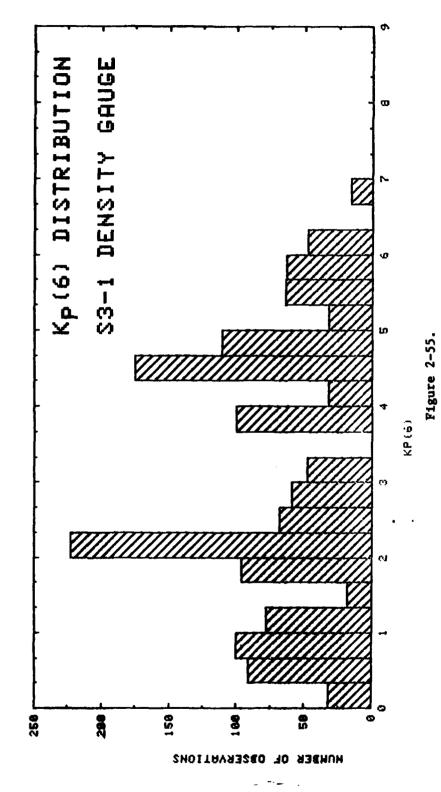
Figure 2-50.



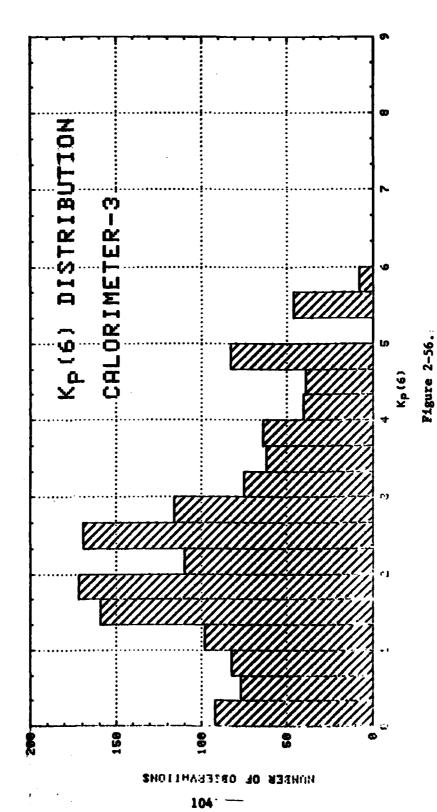


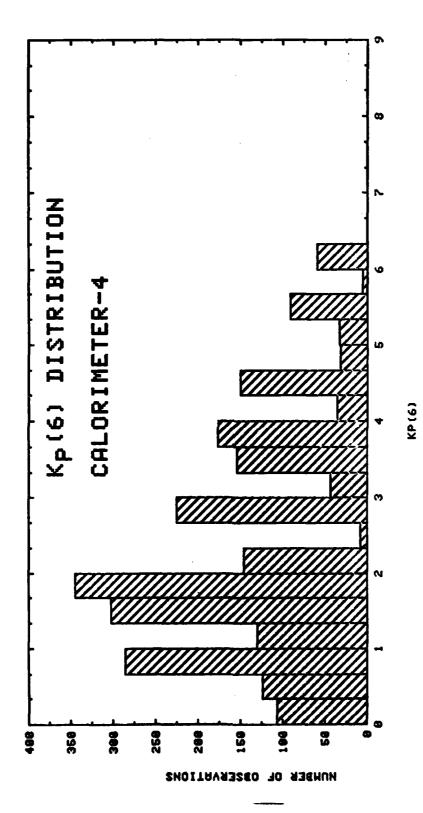


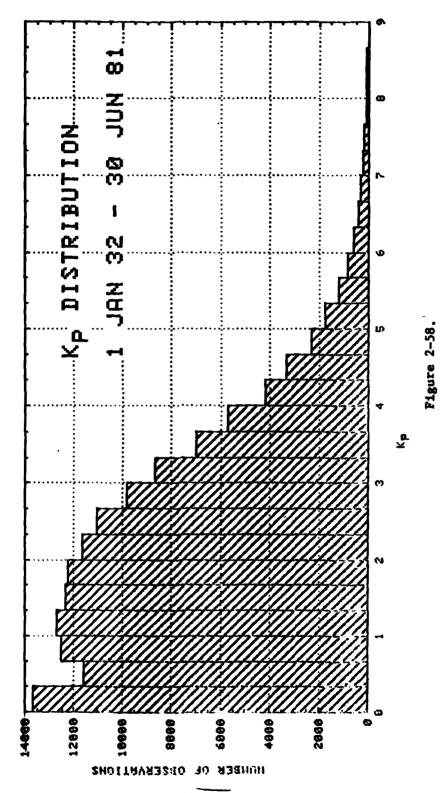




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SECTION 3

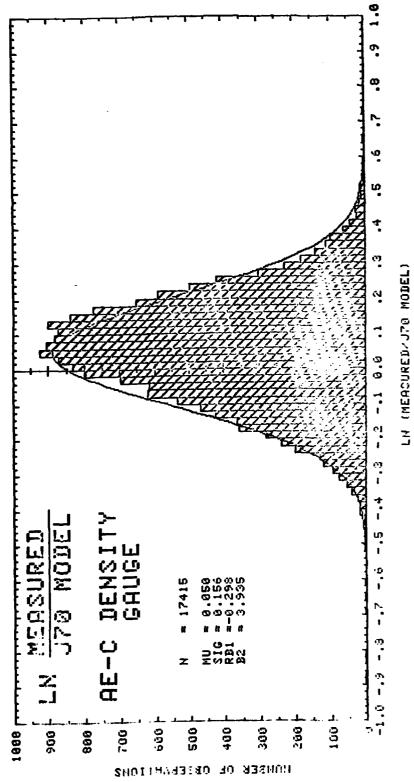


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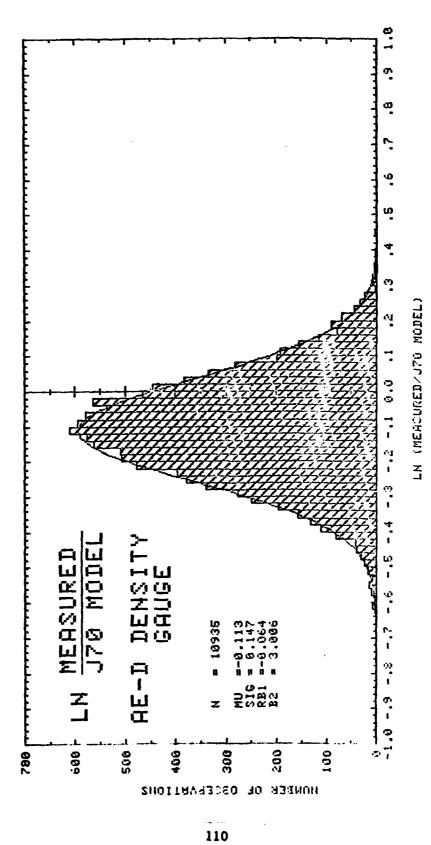
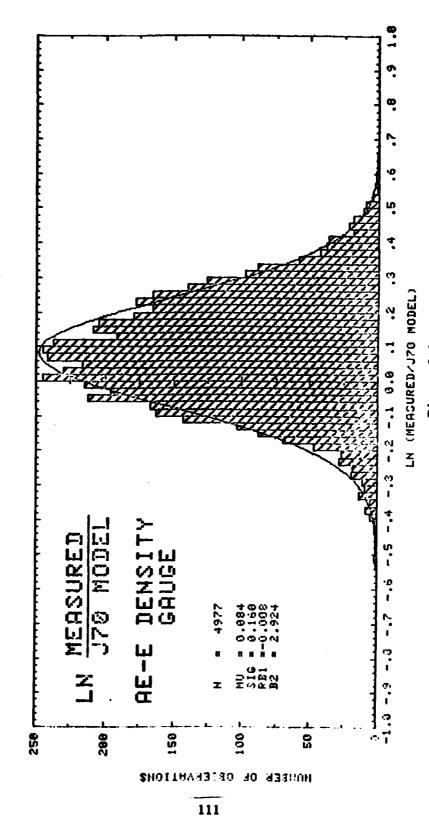


Figure 3-2.



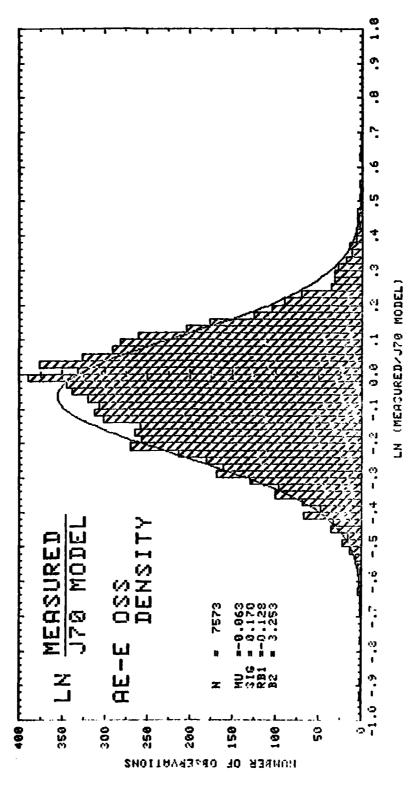


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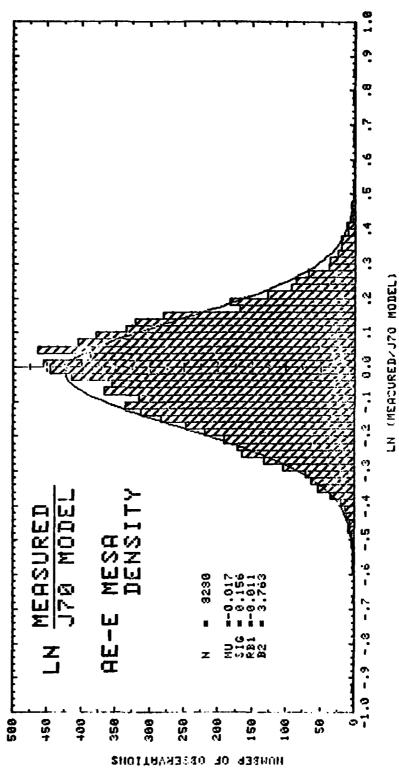
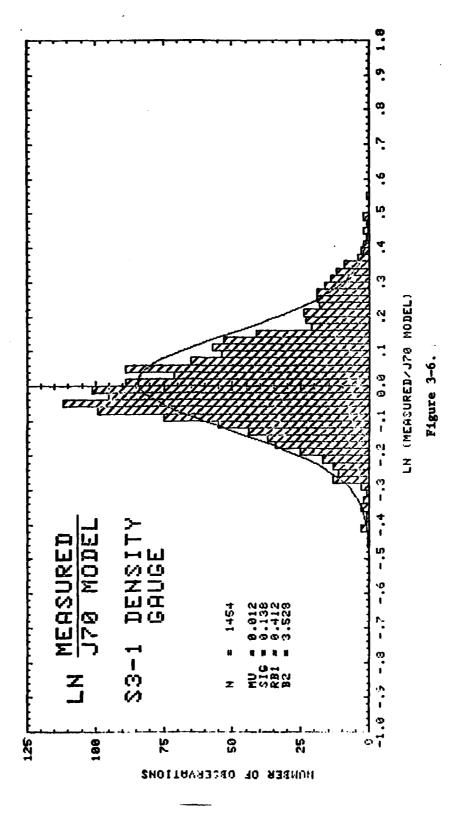
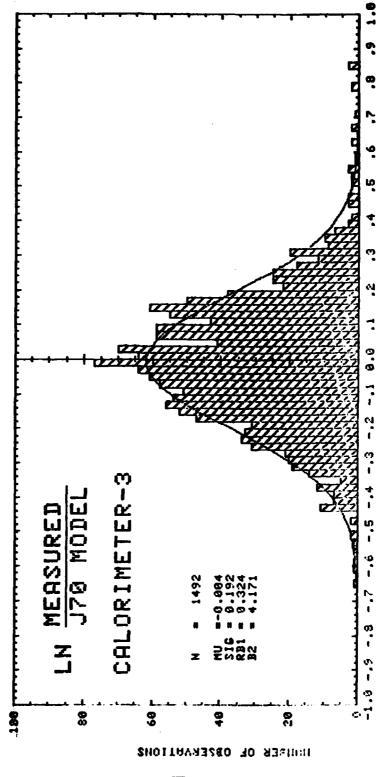


Figure 3-5.

Carrie Contraction





LN (MEASURED/J70 MODEL)

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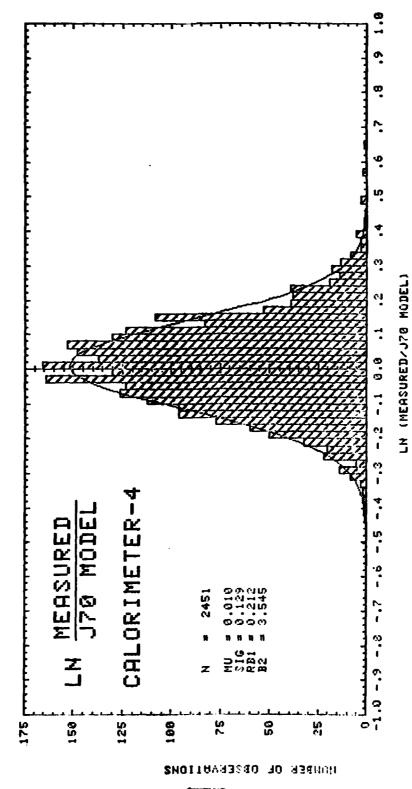


Figure 3-8.

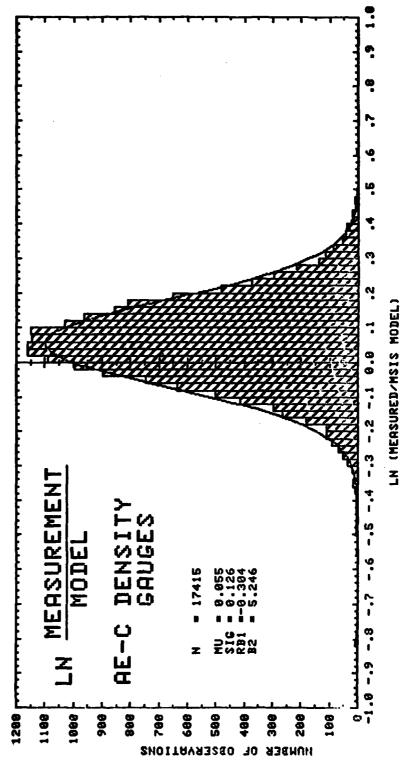


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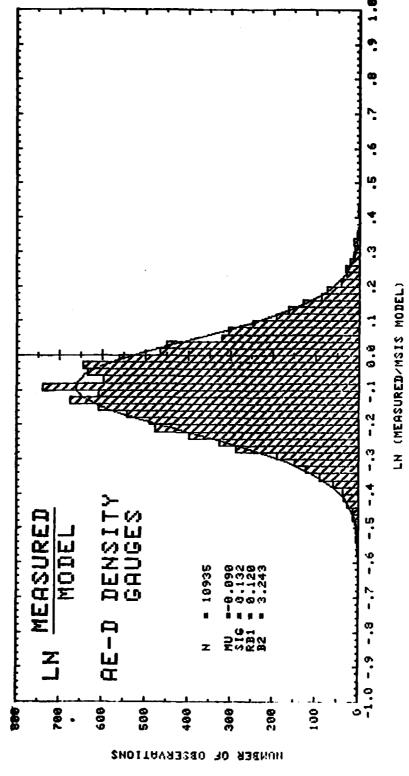


Figure 3-10.

A VALOR

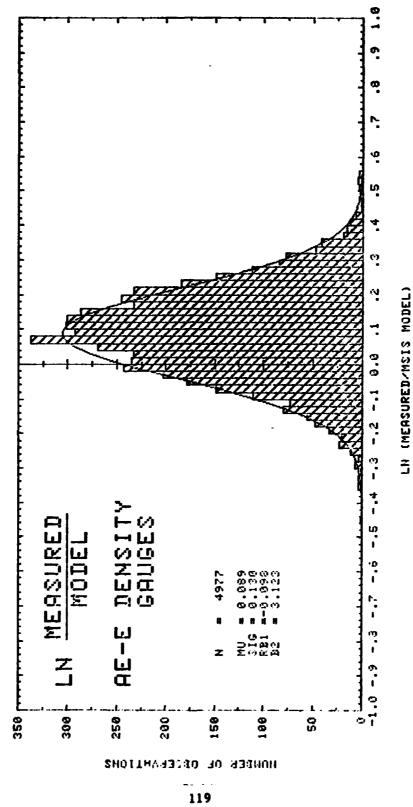
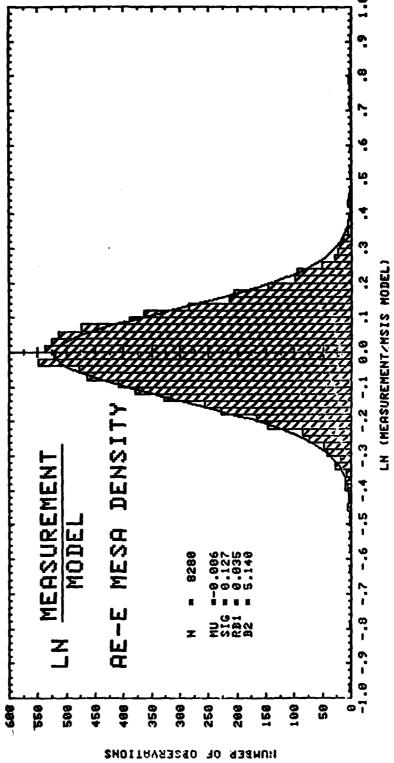


Figure 3-11.

Figure 3-12.



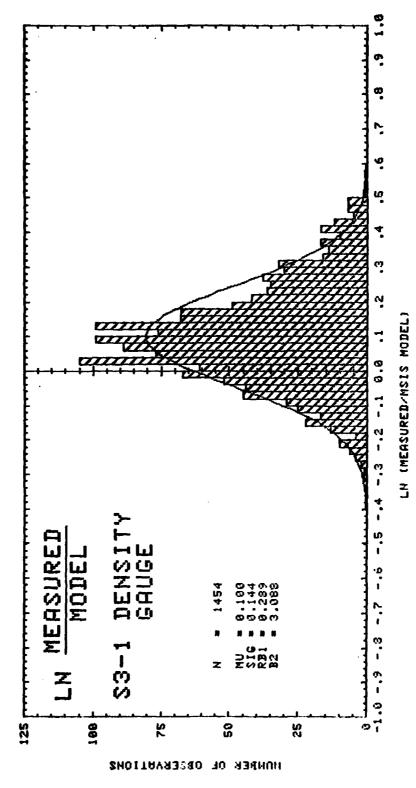
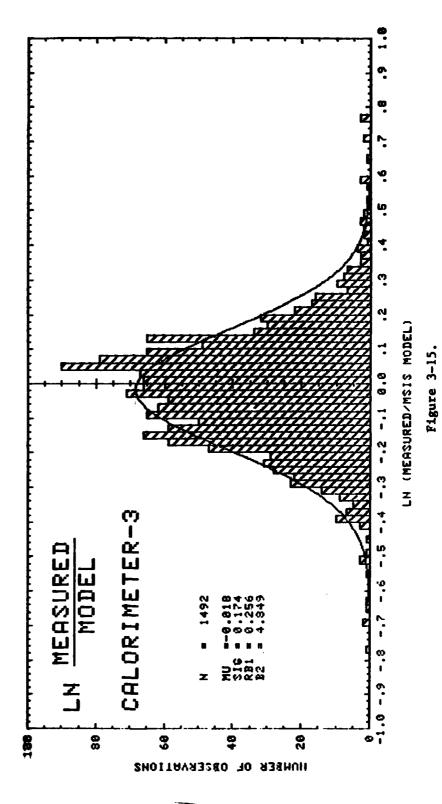


Figure 3-14.



SECTION 4

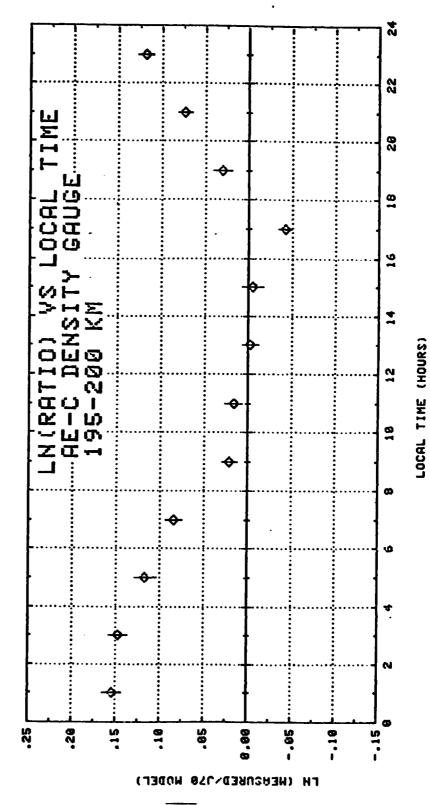


Figure 4-1.

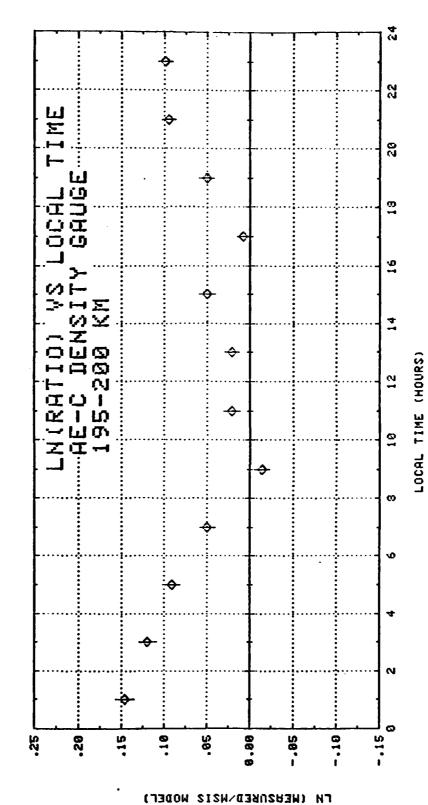
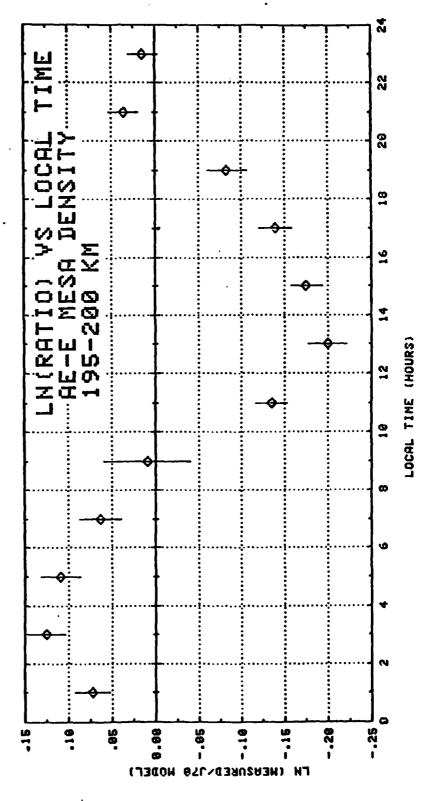
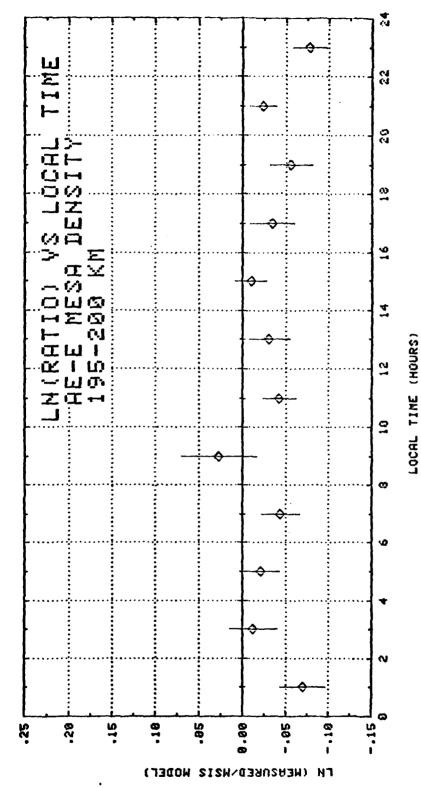


Figure 4-2.





Supplied the supplied to the s

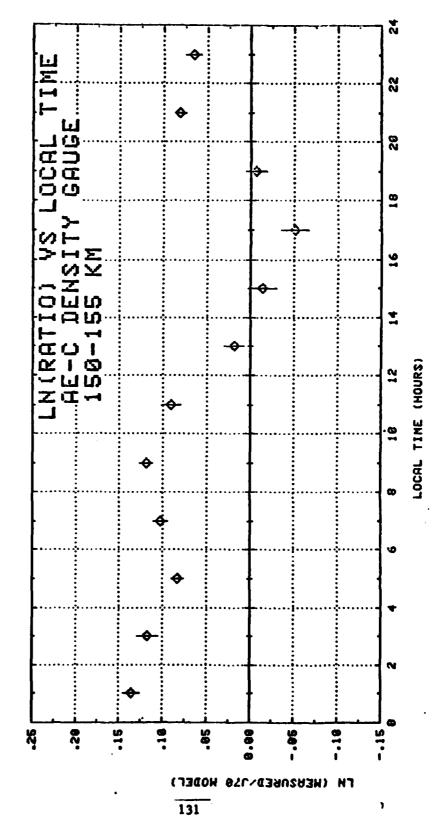


Figure 4-5.

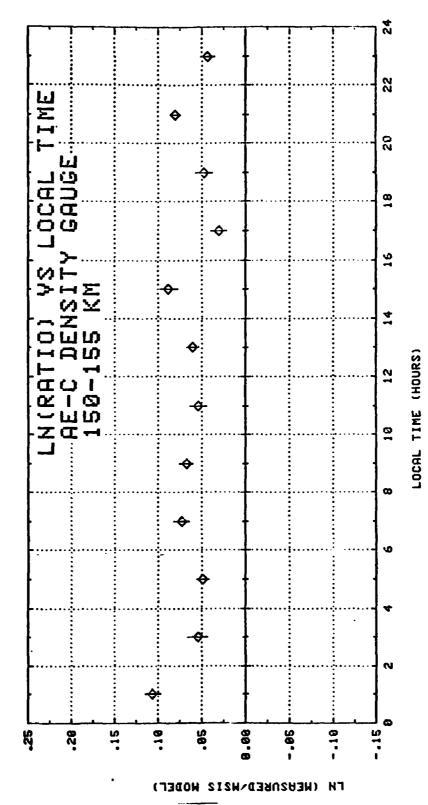


Figure 4-6.

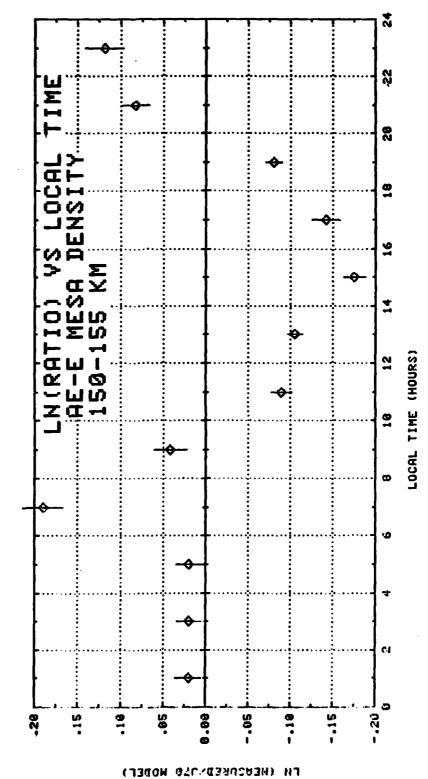


Figure 4-7.

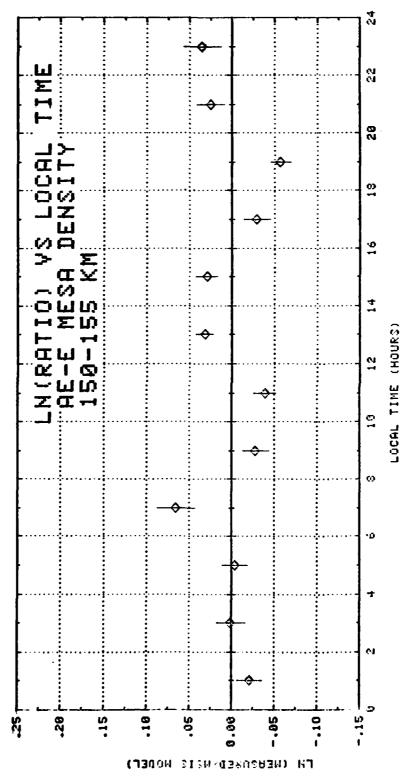


Figure 4-8.

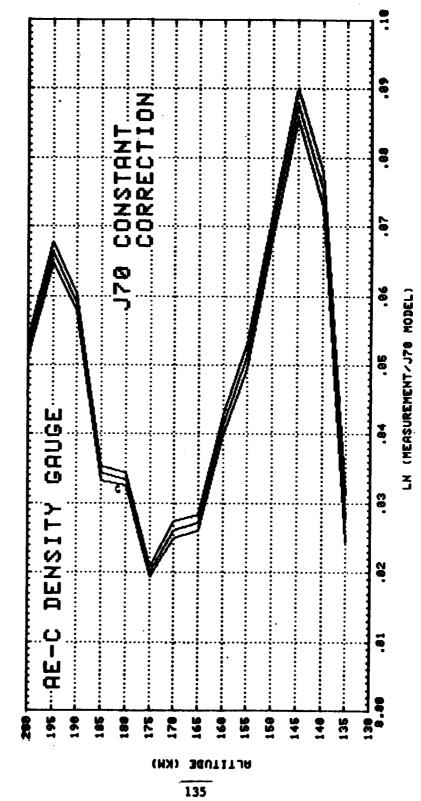


Figure 4-9.

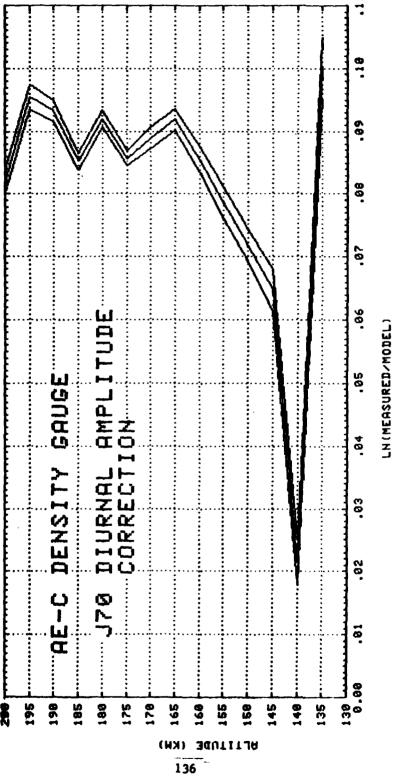


Figure 4-10.

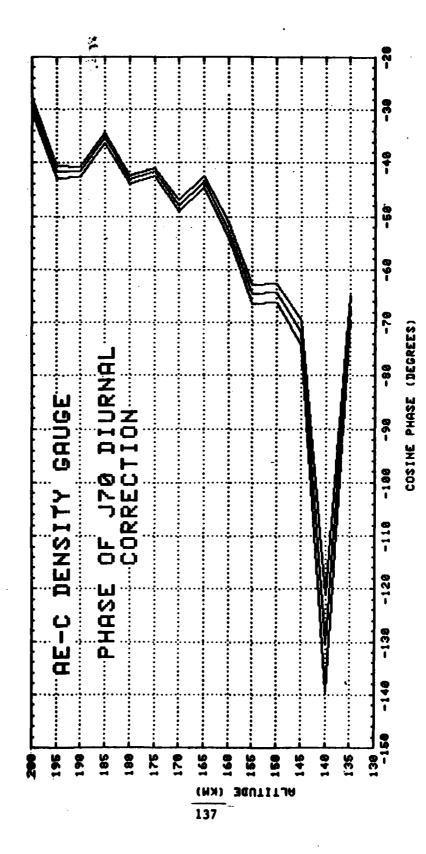


Figure 4-11.

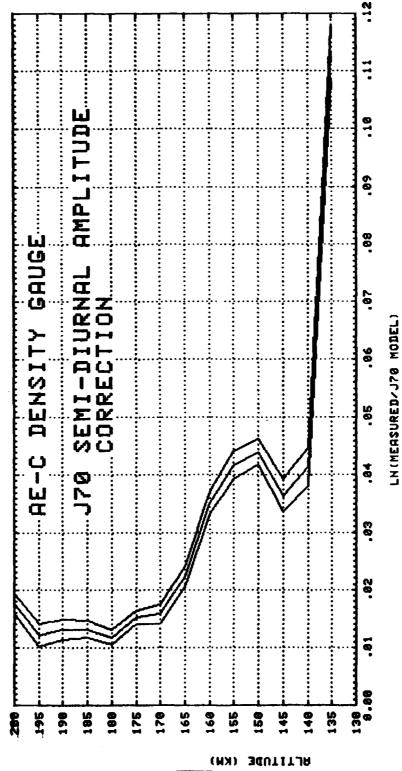
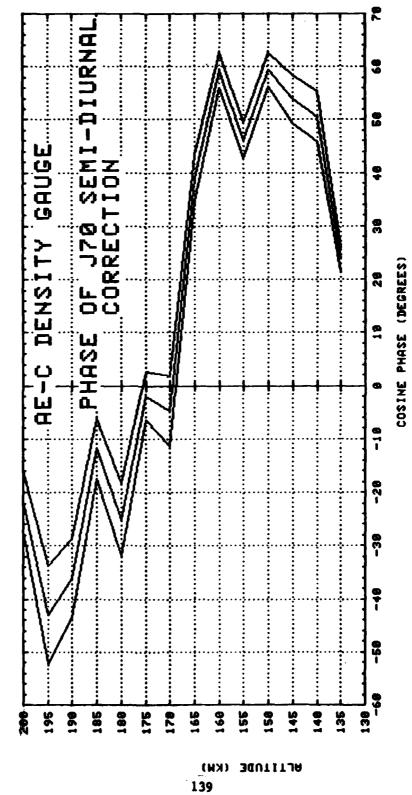


Figure 4-12.

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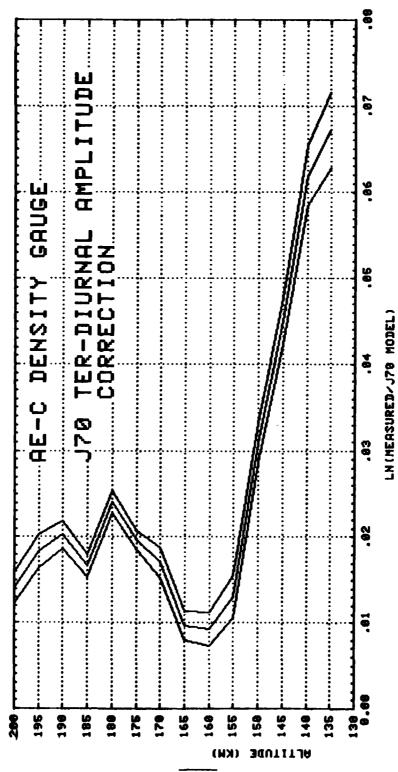


Figure 4-14.

3. V 1.

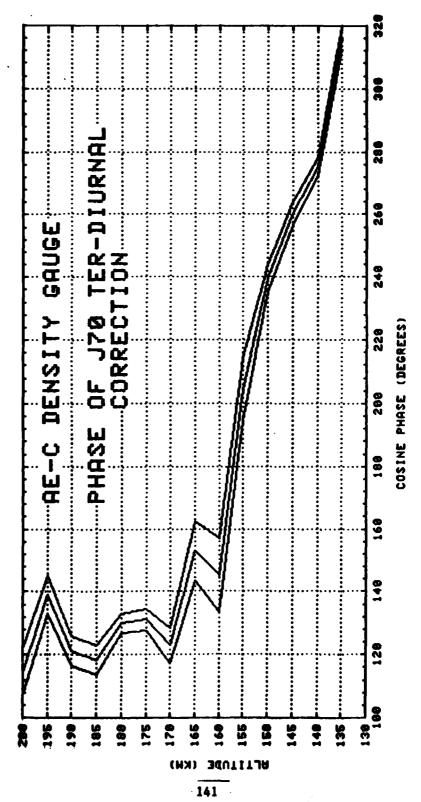
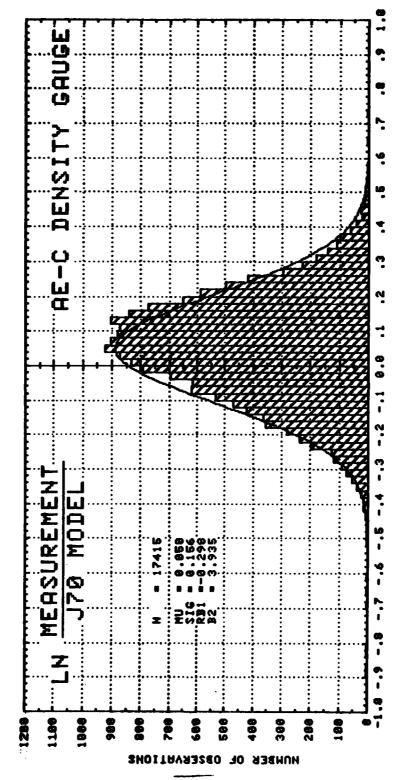
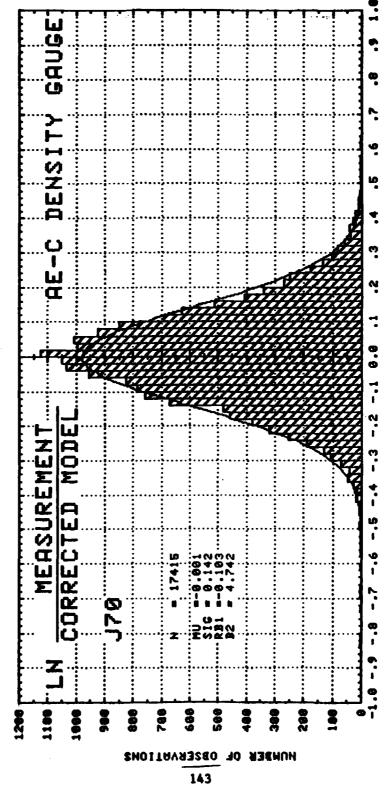


Figure 4-15.



LN(MEASUREMENT/J70 MODEL) Figure 4-16.



LN (MEASUREMENT/CORRECTED J78 MODEL)

Figure 4-17.

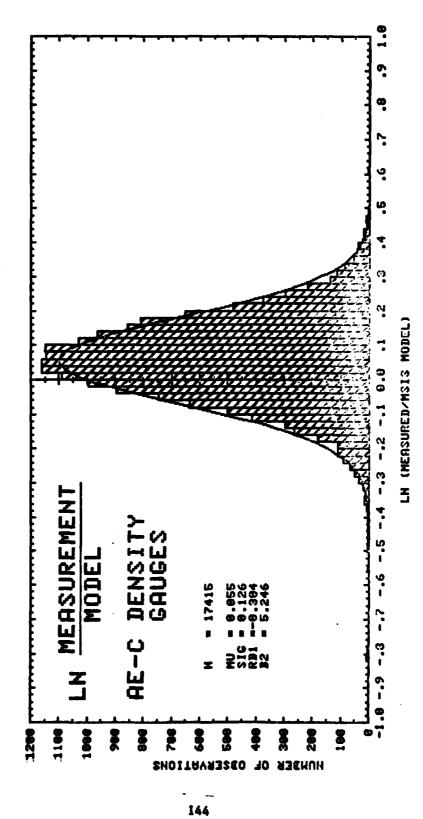


Figure 4-18.

## LABORATORY OFERATIONS

The Laboratory Operations of The Astropace Corporation is conducting emperimental and theoretical investigations necessary for the evaluation and application of accentific advances to now military space systems. Versatility and flexibility have been developed to a high degree by the laboratory personnel in dealing with the many problems encountered in the mation's rapidly developing space systems. Expertise in the latest scientific developments is vital to the accemplishment of tasks related to these problems. The laboratories that contribute to this research are:

Aerophysics Laboratory: Launch vehicle and reentry serodynamics and heat transfer, propulsion chemistry and fluid mechanics, structural mechanics, flight dynamics; high-temperature thermomechanics, gas kinetics and radiation; research in environmental chemistry and contamination; or and pulsed chemical laser development including chemical kinetics, spectroscopy, optical resonators and beam pointing, atmospheric propagation, laser effects and countermeasures.

Chemistry and Physics Laboratory: Atmospheric chemical reactions, atmospheric optics, light scattering, state-specific chemical reactions and radiation transport in rocket plumes, applied laser spectroscopy, laser chemistry, lattery electrochemistry, space vacuum and radiation effects on materials, lubrication and surface phenomena, thermionic emission, photosomositive materials and detectors, atomic frequency standards, and bioenvironmental research and monitoring.

Electronics Research Laboratory: Microelectronics, Gais low-moise and power devices, semiconductor lasers, electronagnetic and optical propagation phenomena, quantum electronics, laser communications, lider, and electro-optics; communication sciences, applied electronics, semiconductor crystal and device physics, rediemetric imaging; millimater-wave and microwave technology.

<u>Information Sciences Research Office</u>: Program verification, program translation, performace-sometrive eyetem design, distributed architectures for epaceborne computers, fault-tolerant computer systems, artificial intelligence, and microelectronics applications.

<u>Materials Sciences Laboratory</u>: Development of new materials: untal untrix cosposites, polymers, and new forms of carbon; component failure smalpsis and reliability; fracture mechanics and stress corrocion; evaluation of unterials in spaces cavironment; materials performance in Space transportation systems; analysis of systems valuerability and survivability in enemy-induced covironments.

Space Sciences Laboratory: Atmospheric and isosopheric physics, rediction from the atmosphere, density and composition of the upper atmosphere, ourses and airgles; megastospheric physics, counic rays, generation and propagation of places were in the megastosphere; solar physics, infrared antronger; the effects of nuclear emplosions, magnetic otores, and solar activity on the earth's atmosphere, isosophere, and magnetosphere; the affects of optical, electromagnetic, and particulate radiations in space on opace systems.

The State of